

IBEX AIRCHARTER
Aircraft Model Hawker HS 125-800XP
Airframe & Engines
Status Report

10 December 2018

Aircraft	MSN / Registration	258319 OD-TSW	Model	HS125-800XP	Manufacturer: Hawker Beechcraft Company			
Certification Date		1997		Current Actual		7,831.07	as @	10-Dec-18
				Total Time		5,219		
				Total Ldgs				
APU	Serial Number	P312	Model	GTCP36-150(W)	Manufacturer Garrett Airesearch			
				TSO	4,885	As @	10-Dec-18	
Engine # 1	Serial Number	P107188	Model	TFE 731-5BR-1H	Manufacturer Honeywell			
				Total Time	7,742	As @	10-Dec-18	
				Total Cycles	5,186			
Engine # 2	Serial Number	P107183	Model	TFE 731-5BR-1H	Manufacturer Honeywell			
				Total Time	7,656	as @	10-Dec-18	
				Total Cycles	5,136			

Forecast

0 Days 0 Hours 0 Landings

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
'05 Time Limits											
050110	B1-B12 Check	B	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
050240	C1-C12 Check	C	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
050370	D1-D12 Check	D	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
050500	E1-E12 Check	E	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
050630	F1-F12 Check	F	24 Months	23-Jul-16	7,523.40	4,866	23-Jul-18	-	-	-140.0	Days
050771	G Check	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
050772	400 Hour Walk-Around Inspection	WAI	400 Flight Hrs	21-Dec-16	7,523.40	4,866	-	7923.40	-	92.3	Hrs
057110	150-250 HOUR ENG INSPECT LH	OOP	250 Eng Hrs	11-Aug-17	7,620.40	5,098	-	7870.40	-	128.7	Eng Hrs
057110	150-250 HOUR ENG INSPECT RH	OOP	250 Eng Hrs	11-Aug-17	7,534.50	5,048	-	7784.50	-	128.7	Eng Hrs
057120	300-400 HOUR ENG INSPECT LH	OOP	400 Eng Hrs	11-Aug-17	7,620.40	5,098	-	8020.40	-	278.7	Eng Hrs
057120	300-400 HOUR ENG INSPECT RH	OOP	400 Eng Hrs	11-Aug-17	7,534.50	5,048	-	7934.50	-	278.7	Eng Hrs
057140	500-850 HOUR ENG INSPECT LH	OOP	850 Eng Hrs	21-Dec-16	7,434.10	4,833	-	8284.10	-	542.4	Eng Hrs
057140	500-850 HOUR ENG INSPECT RH	OOP	850 Eng Hrs	21-Dec-16	7,348.10	4,783	-	8198.10	-	542.3	Eng Hrs
057160	1000-1400 HOUR ENG INSPECT, 72-00-00 P311 LH	OOP	1400 Eng Hrs	18-Aug-15	6,936.70	4,240	-	8336.70	-	595.0	Eng Hrs
057160	1000-1400 HOUR ENG INSPECT, 72-00-00 P311 RH	OOP	1400 Eng Hrs	18-Aug-15	6,850.80	4,191	-	8250.80	-	595.0	Eng Hrs
057170	1200-1650 HOUR ENG INSPECT LH	OOP	1650 Eng Hrs	11-Aug-17	7,620.40	5,098	-	9270.40	-	1528.7	Eng Hrs
057170	1200-1650 HOUR ENG INSPECT RH	OOP	1650 Eng Hrs	20-Aug-15	6,892.00	4,329	-	8542.00	-	886.2	Eng Hrs
72	ACCESSORY GEARBOX LH	OOP	4200 Eng Hrs	20-Dec-05	3,854.60	1,857	-	8254.60	-	512.9	Eng Hrs
72	ACCESSORY GEARBOX RH	OOP	4200 Eng Hrs	19-Aug-16	7,348.10	4,783	-	11748.10	-	4092.3	Eng Hrs
72	Engine Major Periodic InspectionLH	OOP	2100 Eng Hrs	10-Oct-11	5,979.00	3,320	-	8289.00	-	547.3	Eng Hrs
72	Engine Major Periodic InspectionRH	OOP	2100 Eng Hrs	19-Aug-16	7,348.10	4,783	-	9658.10	-	2002.3	Eng Hrs
72	Engine Combustion Plenum Case InspectionLH	OOP	2100 Eng Hrs	10-Oct-11	5,979.00	3,320	-	8289.00	-	547.3	Eng Hrs
72	Engine Combustion Plenum Case InspectionRH	OOP	2100 Eng Hrs	19-Aug-16	7,348.10	4,783	-	9658.10	-	2002.3	Eng Hrs
72	Engine Compressor Zone InspectionLH	OOP	4200 Eng Hrs	20-Dec-05	3,854.60	1,857	-	8254.60	-	512.9	Eng Hrs
72	Engine Compressor Zone InspectionRH	OOP	4200 Eng Hrs	19-Aug-16	7,348.10	4,783	-	11748.10	-	4092.3	Eng Hrs
99	C of A Renewal	OOP	12 Months	4-Mar-17	-	-	4-Mar-18	-	-	-181.0	Days
99	Weight & Balance	OOP	60 Months	2-Feb-14	6,475.70	3,622	1-Feb-19	-	-	53.0	Days
99	HMU Test	OOP	24 Months	4-Dec-15	-	-	3-Dec-17	-	-	-372.0	Days
12 Lubrication											
120001	200 hour lubrication, refer to Part 7 - Lubrication.	200	200 Flight Hrs	11-Aug-17	7,709.80	5,131	-	7909.80	-	78.7	Hrs
120003	400 hour lubrication, refer to Part 7 -Lubrication.	400	400 Flight Hrs	21-Dec-16	7,523.40	4,866	-	7923.40	-	92.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
120020	1600 hour lubrication, refer to Part 7 - LUBRICATION.	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
120030	3200 hour lubrication, refer to Part 7 - LUBRICATION.	D4	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
120049	24 month lubrication, refer to Part 7 - LUBRICATION.	F3	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
120059	48 Month lubrication, refer to Part 7 - LUBRICATION.	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
12	Thrust Control Teleflex Cable, refer to Part 7 - LUBRICATION.	LUB	2100 Eng Hours	17-Jun-11	5,821.00	-	-	7921.00	-	179.3	Eng Hrs
12	Thrust Control Teleflex Cable, refer to Part 7 - LUBRICATION.	LUB	2100 Eng Hours	21-Dec-16	7,523.40	4,866	-	9623.40	-	1967.6	Eng Hrs
20	Standard Practice										
200001	Control surfaces. Do a bonding check (125/H-20, 20-10-5016).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
200002	Refuel point earthing studs (3 off). Do a function test for bonding resistance to the local structure (125/H-20, 20-10-5016).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
200003	Radio racking, electrical-bonding. Do a bonding check with all the	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
21	Air Conditioning										
210002	Ram air valve and dump valve. Do a function test (AMM 21-10-155).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
210003	Environmental control system. Do a function test as follows: • Pressure regulator and shut off valve (PRSOV) (AMM 21-10-76). • PRSOV Overpressure switch (AMM 21-10-78).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
210004	Manual cabin altitude control valve. Make sure of full and free movement.	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
210005	Cooling turbine exhaust duct drain. Make sure the drain is free from blockages.	B7	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
210006	Every 400 hours. Water separator. • Inspect for satisfactory condition (AMM 21-10-95).	OOP	400 Flight Hrs	21-Dec-16	7,523.40	4,866	-	7923.40	-	92.3	Hrs
210007	Every 400 hours. Water separator. Inspect the following items are free from blockages and obstructions (AMM 21-10-95): • Water separator water drain. • Water separator water drain pipeline. • HP line to water injector nozzle. • Water injector nozzle	OOP	400 Flight Hrs	21-Dec-16	7,523.40	4,866	-	7923.40	-	92.3	Hrs
210008	Cabin pressure outflow/safety valve controller. Remove and clean the filter element.	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
210009	Pressurization control venturi screen. Make sure screen is clean.	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
210010	Pressure regulators. Clean the filter screens (AMM 21-30-55).	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
210011	Outflow/safety valves. • Clean the outflow valve poppet valve and seat (AMM 21-30-25) • If required, clean exterior of the outflow valve using a cloth or sponge soaked in cleaning • Check that the orifice in the cabin air filter connection is free of obstruction. fluid (AMM 21-30-25). • Check that the cabin air sense ports are free and clear of any obstruction. • Functionally test operation of valves (AMM 21-30-25, 501).	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
210012	Air conditioning ducts (AMM 21-10-15): • Inspect for satisfactory condition	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
210013	Ram air duct. Inspect, as far as is possible, for cracks in the vicinity of the take-off duct to the shut-off valve, and around the cut-out for the inward relief door.	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
210014	Cabin air filters (pressurization system): • Inspect.	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
210015	Silencer duct and soundproofing (AMM 21-10-15). • Inspect the silencer duct internally for deterioration of the soundproofing material and of the perforated inner retaining duct. • Inspect for internal and external corrosion	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
210016	System overheat temperature switch. Do a bench check (AMM 21-60-85).	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
210017	Baggage pannier heating system inlet and outlet valve • Make sure of the correct operation under normal and "smoke detected" conditions (AMM 21-10-205).	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
210018	Altitude switch, cabin depressurization warning. • Do a functional test (Ref.AMM 21-30-70, 201 (select: 750, 800XP, 800XP/850XP-Pro Line, 900XP)).	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
210019	Cabin supply non-return valve. Inspect, paying particular attention to the security of the carrier arm/butterfly flap rivets (AMM 21-10-107).	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
210020	Electrically actuated auxiliary heat valve. • Do a functional test(AMM 21-60-105 (select: 750, 800XP, 800XP/850XP-Pro Line, 900XP)). NOTE:This test may be done without engine bleed air..	C1	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
210021	APU supply, auxiliary supply, ram air supply and main engine supply. Check/non-return valves: • Remove and inspect. • Make sure that the axial and radial play in the flappers is not excessive (0.025 in. maximum for AiResearch valves)..	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
210024	Low Limit Temperature Control Valve Filter (if installed). • Clean or replace (AMM 21-60-61, 201). NOTE This task is only applicable to airplanes with a HoneywellGTCP36-150W APU, installed to STC ST00459NY	B7	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
210025	APU bleed air system. • Inspect for air leaks at clamped joints and pipes. • Inspect load control valve for cracks, dents, deformation or other damage that may affect the performance of the valve. • Remove and inspect the check valve. Repair as required.	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
21-COMPT	Duct Temperature Switch LH	Bench Chk	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
21-COMPT	Duct Temperature Switch RH	Bench Chk	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
21-COMPT	Main Supply Non Return Valve	O/H	12000 Flight Hrs	27-Jul-97	-	-	-	12000.00	-	4168.9	Hrs
23	Communication										
23-COMPT	Cockpit Voice Recorder	O/H	8000 Flight Hrs	27-Jul-97	-	-	-	8000.00	-	168.9	Hrs
237015	Cockpit Voice Recorder	Inspect	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
23-COMPT	Underwater Locator Beacon Batt-Cvr	Op Chk	24 Months	22-Dec-15	7,194.90	4,517	21-Dec-17	-	-	-354.0	Days
23-COMPT	Underwater Locator Beacon Batt-Cvr	Replace	72 Months	2-Feb-14	6,475.70	3,622	1-Feb-20	-	-	418.0	Days
24	Electrical Power										
240001	Between 200 and 400 Hours. Starter/generators (LH): • Remove and inspect (AMM 24-30-21). • Do a brush wear check (AMM 24-30-21). • Inspect cooling fan. • Inspect splines for wear (AMM 24-30-21). • Inspect engine drive for wear. • Install starter/generator (AMM 24-30-21).	OOP	400 Flight Hrs	21-Dec-16	7,523.40	4,866	-	7923.40	-	92.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
240001	Between 200 and 400 Hours. Starter/generators (RH): <ul style="list-style-type: none"> Remove and inspect (AMM 24-30-21). Do a brush wear check (AMM 24-30-21). Inspect cooling fan. Inspect splines for wear (AMM 24-30-21). Inspect engine drive for wear. Install starter/generator (AMM 24-30-21). 	OOP	400 Flight Hrs	21-Dec-16	7,523.40	4,866	-	7923.40	-	92.3	Hrs
240002	No.1 and No.2 main static inverters. Operate inverters and make sure, by feel of hand, of a positive airflow output from the inverter cooling fans.	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
240003	Frequency wild A.C. system. Do a function test (AMM 24-22-02).	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240004	Busbar contactor control. <ul style="list-style-type: none"> Set the bustie switch to CLOSE. Trip the LH generator off-line. Make sure all d.c. busbars are energised. Reset the LH generator. Trip the RH generator off-line. Make sure all d.c. busbars are energised. Reset the RH generator 	C1	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
240005	APU generator overheat indication circuit. Function test (AMM 24-36-03, Turbomach APU installation only).	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240006	D.C. busbar protection/voltage sensing unit. Check the trip voltage datums (AMM 24-30-142).	D6	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240007	Ground power overvoltage sensor. Bench check the trip voltage datum (AMM 24-30-146).	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240008	Task deleted.	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240009	D.C. generating system. Do a function test of the emergency contactor control (AMM 24-30-01).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
240010	PE, PS1 and PS2 busbar feeder cables below Panel DA. <ul style="list-style-type: none"> Inspect for discoloration due to overheating. Torque tighten the connections (AMM 24-01-00).. 	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
240011	Busbar PE, PS1 and PS2 feeder fuses. Replace (with new items) the fuses: <ul style="list-style-type: none"> F31 an F32 on panel ZL. F13, F14, F23 and F24 on panel GA When you do this: Make sure the fuses are in correct contact with the busbars and the mounting pillars. Make sure the fuses are clamped in position correctly. Torque tighten the securing nuts (AMM 24-01-00). 	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
240012	Busbar PE, PS1 and PS2 feeder system connections: <ul style="list-style-type: none"> Inspect for discoloration due to overheating. Torque tighten the connections (AMM 24-01-00). 	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
240013	All circuit breakers. Function test for correct mechanical operation. Do two manual switching cycles for this test.	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240014	Main generator control unit (GCU). Do a function test of the overvoltage protection circuit (AMM 24-30-07).	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240015	APU Generator control unit (GCU). Do a function test of the overvoltage protection circuit (AMM 24-36-03).	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240016	Electrical cable pressure bungs (in the forward pressure bulkhead). Inspect for correct location in their housings (AMM 24-01-00).	D3	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
240017	Earth stations *16, *17, *18 and *19: • Inspect. Look for signs of arcing. • Do an impedance check (AMM 24-01-00).	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
240018	Electrical cable pressure bungs (in the rear pressure-bulkhead). Inspect for correct location in their housings (AMM 24-01-00).	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240024	Every 600 APU running hours. APU Starter/Generator • Remove (AMM 24-30-21). • Check the brush wear (AMM 24-30-21). • Inspect the cooling fan. • Inspect the splines for wear (AMM 24-30-21). • Inspect the engine drive for wear. Install the APU starter/generator (AMM 24-30-21).	OOP	600 APU Hrs	24-Jul-14	6,626.40	3,798	-	7226.40	-	2341.4	APU Hrs
240025	Electrical cable pressure bungs (in the fuselage keel at the front spar). Inspect for correct location in their housings (AMM 24-01-00).	D5	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240026	Electrical cable pressure bungs (in the forward pressure bulkhead). Inspect for correct location in their housings (AMM 24-01-00).	D4	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
240027	Main batteries Rebling Plastics "Quick Disconnect" battery connectors, P/N's 7007, 7013, 7016 and 7017, • 800XP airplanes, Check (AMM 24-30-104, 401 or 24-30-151, 401)	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
24-COMPT	Starter Generator RH	O/H	1000 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8523.40	-	692.3	Hrs
24-COMPT	Starter Generator LH	O/H	1000 Flight Hrs	18-Aug-15	7,067.50	4,326	-	8067.50	-	236.4	Hrs
24-COMPT	Frequency Wild Ac Generator LH	O/H	96 Months	17-Jun-11	5,911.20	-	15-Jun-19	-	-	187.0	Days
24-COMPT	Frequency Wild Ac Generator LH	O/H	3000 Flight Hrs	17-Jun-11	5,911.20	-	-	8911.20	-	1080.1	Hrs
24-COMPT	Frequency Wild Ac Generator RH	O/H	96 Months	17-Jun-11	5,911.20	-	15-Jun-19	-	-	187.0	Days
24-COMPT	Frequency Wild Ac Generator RH	O/H	3000 Flight Hrs	17-Jun-11	5,911.20	-	-	8911.20	-	1080.1	Hrs
24-COMPT	Bus Tie Contactor	O/H	5000 Flight Hrs	28-Feb-08	4,735.10	2,268	-	9735.10	-	1904.0	Hrs
24-COMPT	Inverter Supply Contactor #1	O/H	5000 Flight Hrs	28-Feb-08	4,735.10	2,268	-	9735.10	-	1904.0	Hrs
24-COMPT	Inverter Supply Contactor #2	O/H	5000 Flight Hrs	28-Feb-08	4,735.10	2,268	-	9735.10	-	1904.0	Hrs
24-COMPT	Ground Supply Start Contactor	O/H	6000 Flight Hrs	4-Jun-10	5,543.70	-	-	11543.70	-	3712.6	Hrs
243154	Battery, Main #1	CAP Chk	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
243154	Battery, Main #1	Replace	4 Years	19-Aug-16	7,523.40	4,866	18-Aug-20	-	-	617.0	Days
243155	Battery, Main #2	CAP Chk	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
243155	Battery, Main #2	Replace	4 Years	20-Aug-15	7,068.50	4,329	19-Aug-19	-	-	252.0	Days
243156	Battery #3	CAP Chk	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
243156	Battery #3	CAP Chk	300 Flight Hrs	11-Aug-17	7,709.80	5,131	-	8009.80	-	178.7	Hrs
243157	Battery #4	CAP Chk	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
243157	Battery #4	CAP Chk	300 Flight Hrs	11-Aug-17	7,709.80	5,131	-	8009.80	-	178.7	Hrs
243158	Battery #5	CAP Chk	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
243158	Battery #5	CAP Chk	300 Flight Hrs	11-Aug-17	7,709.80	5,131	-	8009.80	-	178.7	Hrs
24-COMPT	Davtron Clock Battery LH	Replace	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
24-COMPT	Davtron Clock Battery RH	Replace	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
25	Equip / Furnishings										
250003	3rd crew members seat: • Make sure that the seat stowing action is satisfactory. • Make sure of correct operation and satisfactory condition.	B6	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
250005	Emergency packs . Inspect contents for condition and make sure expiry dates are valid.	E6	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
250006	First aid kits. Make sure expiry date is valid and make sure seal is intact.	E6	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
250007	Crew seats. • Do an operational check (Ref. CMM 25-11-24, 501 (Check1)) or (CMM 25-11-27, 501 (Check 1))..	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
250008	Crew seats. • Remove the crew seats (Ref. AMM 25-10-14, 401) • Do a detailed check (Ref. CMM 25-11-24, 501 (Check 2)) or (ref. CMM 25-11-67, 501(Check 2)). • Install the crew seats (Ref. AMM 25-10-14, 401).	D4	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
25-COMPT	Elt	Inspect	24 Months	22-Dec-15	7,194.90	4,517	21-Dec-17	-	-	-354.0	Days
256010	Elt, Functional Check	F CHK	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
25-COMPT	Elt	O/H	60 Months	2-Feb-14	6,475.70	3,622	1-Feb-19	-	-	53.0	Days
25-COMPT	Life Vest Pax #1	O/H	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	Life Vest Pax #2	O/H	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	Life Vest Pax #3	O/H	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	Life Vest Pax #4	O/H	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	Life Vest Pax #5	O/H	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	Life Vest Pax #6	Inspect	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	Life Vest Pax #7	Bench Chk	60 Months	24-May-14	-	-	23-May-19	-	-	164.0	Days
25-COMPT	PBE	Remove	120 Months	3-Jan-12	6,196.00	-	31-Dec-21	-	-	1117.0	Days
25-COMPT	Life Vest Pilot	Bench Chk	60 Months	9-Jun-16	7,438.80	4,775	8-Jun-21	-	-	911.0	Days
25-COMPT	Life Vest Co-Pilot	Bench Chk	60 Months	9-Jun-16	7,438.80	4,775	8-Jun-21	-	-	911.0	Days
25-COMPT	First Aid Kit	Bench Chk	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
25-SB	Appliance	SB	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
25-SB	Appliance	SB	2400 Hrs	19-Aug-16	7,523.40	4,866	-	9923.40	-	2092.3	Hrs
26	Fire Protection										
260001	Airplanes with Fire Extinguisher Pressure Gages ONLY. Fire extinguishers, main engines. • Check the pressure gage. Make sure the reading is correct, refer to the pressure/temperature chart. • Do a functional test (AMM 26-20-00). NOTE For airplanes without Fire Extinguisher Pressure gages refer to Inspection E7 task 260005.	E1	12 Months	9-Aug-13	6,384.70	N/A	9-Aug-14	-	-	N/A	N/A
260002	Rear equipment bay overheat detection system. Do a functional test (26-10-00).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
260004	Portable fire extinguisher: (C/P) • Check for correct weight or if gauge is fitted, make sure the needle is in the green sector. • Inspect and look for signs of leaks or discharge.	E6	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
260005	For airplanes without Fire Extinguisher Pressure Gages ONLY. Fire extinguishers, main engines: • Remove (AMM 26-20-12). • Inspect. • Check weight (AMM 26-20-12). • Install (AMM 26-20-12). • Do a functional test (AMM 26-20-00). NOTE For airplanes with Fire Extinguisher Pressure Gages refer to Inspection E1, task 260001.	E7	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
260007	Fire extinguisher, APU (Other than PATS APU GTCP36-150W): • Inspect. • Check pressure gauge (AMM 26-20-54). • Check Hydrostatic and Cartridge due dates. • Do a functional test (AMM 26-20-00).	N/A								N/A	N/A
260008	Every 6 months. Smoke Detection System : • Do a function test: 800XP and 800XP Pro Line 21 airplanes (MMA 26-11-00).	OOP	6 Months	11-Aug-17	7,709.80	5,131	9-Feb-18	-	-	-303.5	Days
260010	PATS APU GTCP36-150W. APU fire extinguisher. • Inspect. • Check pressure gauge (Meggitt CMM 26-20-02). • Check Hydrostatic and Cartridge due dates.	E7	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
26-COMPT	Overheat Detector Switch #1	Replace	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	Overheat Detector Switch #2	Replace	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	Overheat Detector Switch #3	Replace	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	Overheat Detector Switch #4	HPT	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	Overheat Detector Switch #5	HPT	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	Overheat Detector Switch #6	O/H	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	APU Generator overheat Thermal Switch	HPT	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	APU Fire Warning Switch	ReCharge	60 Months	21-Dec-12	6,237.40	3,544	20-Dec-17	-	-	-355.0	Days
26-COMPT	APU Fire Ext Cartridge	Replace	120 Months	30-Nov-08	5,345.40	-	28-Nov-18	-	-	-12.0	Days
26-COMPT	APU Fire Ext Cartridge Manuf Date	Replace	180 Months	30-Nov-08	5,345.40	1,857	27-Nov-23	-	-	1813.0	Days
26-COMPT	Engine Fire Ext Cartridge AFT HEAD B	O/H	120 Months	4-Nov-11	6,068.40	-	1-Nov-21	-	-	1057.0	Days
26-COMPT	Engine Fire Ext Cartridge Manuf Date AFT HEAD A	Replace	180 Months	4-Nov-11	6,068.40	-	31-Oct-26	-	-	2882.0	Days
26-COMPT	Engine Fire Ext Cartridge Manuf Date AFT HEAD B	Replace	180 Months	4-Nov-11	6,068.40	-	31-Oct-26	-	-	2882.0	Days
26-COMPT	Engine Fire Ext Cartridge Manuf Date FWD HEAD A	ReCharge	180 Months	4-Nov-11	6,068.40	-	31-Oct-26	-	-	2882.0	Days
26-COMPT	Engine Fire Ext Cartridge Manuf Date FWD HEAD B	HST	180 Months	4-Nov-11	6,068.40	-	31-Oct-26	-	-	2882.0	Days
26-COMPT	Engine Fire Ext Cartridge AFT HEAD A	O/H	120 Months	4-Nov-11	6,068.40	-	1-Nov-21	-	-	1057.0	Days
26-COMPT	Engine Fire Ext Cartridge FWD HEAD A	Replace	120 Months	4-Nov-11	6,068.40	-	1-Nov-21	-	-	1057.0	Days
26-COMPT	Engine Fire Ext Cartridge FWD HEAD B	Replace	120 Months	4-Nov-11	6,068.40	-	1-Nov-21	-	-	1057.0	Days
26-COMPT	Engine Fire Extinguisher LH (AFT)	Replace	120 Months	22-Dec-15	7,194.90	4,517	19-Dec-25	-	-	2566.0	Days
26-COMPT	Engine Fire Extinguisher RH (FWD)	Replace	120 Months	22-Dec-15	7,194.90	4,517	19-Dec-25	-	-	2566.0	Days
26-COMPT	APU Fire Extinguisher	O/H	60 Months	20-Dec-14	6,756.42	3,974	19-Dec-19	-	-	374.0	Days
26-COMPT	Cabin Fire Extinguisher	Replace	144 Months	12-Jun-15	7,067.50	4,326	9-Jun-27	-	-	3103.0	Days
26-COMPT	Cabin Fire Extinguisher	HPT	72 Months	12-Jun-15	7,067.50	4,326	10-Jun-21	-	-	913.0	Days
26-COMPT	Fire Extinguisher Fired Fuse #1	Replace	36 Months	19-Aug-16	7,523.40	4,866	19-Aug-19	-	-	252.0	Days
26-COMPT	Fire Extinguisher Fired Fuse #2	Replace	36 Months	11-Aug-17	7,709.80	5,131	10-Aug-20	-	-	609.0	Days
26-COMPT	Fire Extinguisher Fired Fuse (MFG Date) #1	O/H	48 Months	20-Dec-14	6,756.42	3,974	19-Dec-18	-	-	9.0	Days
26-COMPT	Cockpit Fire Extinguisher	HPT	72 Months	30-Nov-14	5,345.40	-	28-Nov-20	-	-	719.0	Days
26-COMPT	Cockpit Fire Extinguisher	Replace	144 Months	30-Nov-09	5,345.40	-	27-Nov-21	-	-	1083.0	Days
27	Flight Controls										
270001	Every 400 hours. Airbrake selector lever baulk. Do an operational test (AMM 27-61-03, page block 501). NOTE This task is ONLY applicable to airplanes with Mod.25F743A or 25F743C embodied and is applicable to Canadian registered airplanes only. Mod.25F743A (SB 27-3150) introduces an airbrake selector lever baulk to 800XP airplanes.	OOP	400 Flight Hrs	24-Jul-14	6,626.40	3,798	-	7026.40	-	N/A	N/A
270002	Flap vane inner. With the flaps lowered, inspect the left and right outer attachment bracket for cracking and for security of attachment to the flap leading edge.	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270003	Flap screwjacks. Inspect the screw threads for satisfactory condition.	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
270004	Stall warning and identification system. <ul style="list-style-type: none"> Do a functional test of the stall warning and identification system (AMM 27-33-00). Do an angle verification check (AMM 27-33-00). Do an electrical zero check (AMM 27-33-00). NOTE The electrical zero check is only applicable to airplanes with Mod. 25G206A or Mod. 25G206B.	B2	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270008	Rudder bias strut heater muff. <ul style="list-style-type: none"> Do a functional test of the rudder bias strut heater muff(AMM 27-20-00). 	B2	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270009	Stall detector unit (third channel). Do a function test (AMM 27-33-14). <ul style="list-style-type: none"> 800XP airplanes (AMM 27-33-14). NOTE This task is only applicable to 800XP and 800XP Pro Line 21airplanes Pre Mod. 25G206A or Pre Mod. 25G206A and25G206B.	B2	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270011	Gust lock, ailerons and elevators. <ul style="list-style-type: none"> Inspect. Make sure that the bush in the mounting bracket is lightly lubricated and free from burrs or scores. With the unit in the operated and locked position but without the engagement of the claw spigots in the control column bushes, check for positive spring pressure on the claws and for initial resistance to movement when releasing the knurled locking ring 	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270012	Rudder trim tab and anti-servo tab. <ul style="list-style-type: none"> Check the free movement (backlash) at the trailing edges of the tabs (AMM 27-20-00, 501). While the connecting rods are disconnected check the rear rod-end bearings for smooth operation. Detail inspect the connecting rod attachment lugs on the tabs for cracks using a X 10 magnifier. NOTE: Do not remove the protective coating unless a defect, or suspected defect, is found.	B8	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270013	Elevators and trim tabs. <ul style="list-style-type: none"> Check the free movement (backlash) at the trailing edges of the tabs (AMM 27-30-00, 501). While the connecting rods are disconnected check the rear rod-end bearings for smooth operation. Detail inspect the connecting rod attachment lugs on the trim tabs for cracks, using a X 10 magnifier. NOTE Do not remove the protective coating unless a defect, or suspected defect, is found. Rudder and elevators primary surfaces. Make sure of full and free movement from stop to stop. B8 Aileron primary surfaces. Make sure of full and free movement from stop to stop and that all accessible aileron pulleys rotate freely	B8	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270014	Rudder and elevators primary surfaces. Make sure of full and free movement from stop to stop.	B8	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270015	Aileron primary surfaces. Make sure of full and free movement from stop to stop and that all accessible aileron pulleys rotate freely.	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
270016	<p>Aileron trim and servo tabs.</p> <ul style="list-style-type: none"> • Check the free movement (backlash) at the trailing edges of the tabs (AMM 27-10-00, 501). • While the connecting rods are disconnected inspect the r rod-end bearings for smooth operation. • Detail inspect the connecting rod attachment lugs on the tabs for cracks using a X 10 magnifier. <p>NOTE Do not remove the protective finish unless a defect, or a suspected defect is found.</p>	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270017	<p>Flap Screwjack and reduction gearbox.</p> <ul style="list-style-type: none"> • Check the gearbox fluid level (AMM 27-50-141). 	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270018	<p>Flap synchronization system control cables.</p> <ul style="list-style-type: none"> • Inspect for fraying, particularly in the vicinity of the pulleys. • Make sure accessible pulleys rotate freely. 	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270019	<p>Flap control unit.</p> <ul style="list-style-type: none"> • Check the gearbox fluid level (AMM 27-50-121). 	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270020	<p>Flap control unit.</p> <ul style="list-style-type: none"> • With the hydraulic system depressurized, disconnect the connecting rod at the input lever. • Operate the lever to displace the relay valve from the neutra position. • Release the lever and make sure the relay valve returns to the neutral position. • Reconnect the connecting rod. 	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270021	<p>Flap drive shafts.</p> <p>Inspect for signs of corrosion and for movement between the end fittings and the shafts</p>	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270022	<p>Aileron primary surfaces.</p> <p>Make sure of full and free movement from stop to stop and that all accessible aileron pulleys rotate freely.</p>	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270023	<p>Aileron trim and servo tabs.</p> <ul style="list-style-type: none"> • Check the free movement (backlash) at the trailing edges of the tabs (AMM 27-10-00, 501). • While the connecting rods are disconnected inspect the r rod-end bearings for smooth operation. • Detail inspect the connecting rod attachment lugs on the tabs for cracks using a X 10 magnifier. <p>NOTE Do not remove the protective finish unless a defect, or a suspected defect is found.</p>	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270024	<p>Flap Screwjack and reduction gearbox.</p> <ul style="list-style-type: none"> • Check the gearbox fluid level (AMM 27-50-141) 	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270025	<p>Flap synchronization system control cables.:</p> <ul style="list-style-type: none"> • Inspect for fraying, particularly in the vicinity of the pulleys. • Make sure accessible pulleys rotate freely. 	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
270026	Flap drive shafts. Inspect for signs of corrosion and for movement between the end fittings and the shafts.	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270027	Flap control system. Do a function test (AMM 27-50-02, 501).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270028	Airbrake and lift dump system. • Do a functional test (AMM 27-61-02, 501).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270030	Elevator trim servo. • Operate the elevator trim switch on the left, or right, control column. elevator trim wheel with sufficient • As the elevator trim servo motor operates the elevator trim tabs, hold the manual force to stop the trim wheel. • The force needed to stop the trim wheel must not be more than 15 lbs	C2	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270032	Aileron autopilot servo control cables. Check cable tensions are correct: • 800XP airplanes (AMM 27-00-00, 201 and 27-10-93, 401).	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270033	Elevator control system balance/feel spring. Inspect.	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270034	Elevator primary control cables. Check cable tensions are correct (AMM 27-30-00).	C5	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270035	Flap control cables. Check cable tensions are correct (AMM 27-50-02).	C5	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270036	Elevator primary and elevator trim control cables. Check cable tensions are correct (AMM 27-30-00).	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270037	Rudder primary and rudder trim control cables. Check cable tensions are correct (AMM 27-20-00).	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270038	Elevator autopilot servo control cables. Check cable tensions are correct (AMM 27-30-81, 401)	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270039	Rudder autopilot servo control cables. Check cable tensions are correct: • 800XP airplanes (AMM 27-20-121, 401).	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270040	Elevator autopilot trim-servo control cables. Check cable tensions are correct (AMM 27-30-00).	C0	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270041	Elevator trim cables: • Inspect the condition of the protective coating (AMM 20-95-11, Item No. 079) on the cables over a length of 42 ins (1070 mm) from each trim chain where the cables are exposed between the horizontal stabilizer and the elevators. • Apply protective coating (AMM 27-30-00, page block 301), if necessary	C8	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270042	Aileron primary and trim control cables. Check cable tensions are correct (AMM 27-10-00).	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270043	Flap drive shaft universal joints: • Remove joint gaiters and inspect joints for wear. • Repack with grease (AMM 20-95-11, Item No. 040). • Reinstall joint gaiters.	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270044	Flap Bevel drive assembly. Check that the backlash between the input and output shafts, measured at the input coupling flange bolts, does not exceed 0.020 in.	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270045	Flap control unit baulk release cable. Check cable tension is correct (AMM 27-50-02).	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
270046	Flap/airbrake baulk bobbin. Make sure bobbin setting is correct (AMM 27-50-02).	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270047	Flap spring drum pulley-lever. <ul style="list-style-type: none"> • Disconnect rod from pulley lever. • Check the pull off force required to move the lever off the spring drum internal stops is 5.35 lb. to 7 lb., measured at the connecting rod attachment point on the lever. • Check that the lever returns to contact internal stops at or before 5 lb. force at the connecting rod attachment. • Reconnect rod to pulley lever. 	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270048	Flap Screwjack and reduction gearbox. Check that the backlash on the screwthreads does not exceed 0.020 in (AMM 27-50-141).	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270049	Flap synchronising system: <ul style="list-style-type: none"> • Check the cable tensions are correct (AMM 27-50-02). • With the flaps selected up, disconnect the operating link from one transmitter gearbox only. • Pressurize the hydraulic system. • Whilst restraining the disconnected lever arm, select the flaps down. • Make sure that the flaps start to move and then the system cuts out. • Depressurize the hydraulic system and connect the transmitter operating link. • Pressurize the hydraulic system and make sure that the flaps operate correctly. 	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270050	Airbrake control cables. Check the cable tensions are correct (AMM 27-61-02).	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270051	Aileron primary and trim control cables. Check cable tensions are correct (AMM 27-10-00).	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270052	Flap drive shaft universal joints: <ul style="list-style-type: none"> • Remove joint gaiters and inspect joints for wear. • Repack with grease (AMM 20-95-11, Item No. 040). • Reinstall joint gaiters 	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270053	Flap bevel drive assembly. Check that the backlash between the input and output shafts, measured at the input coupling flange bolts, does not exceed 0.020 in.	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270054	Flap screwjack and reduction gearbox. Check that the backlash on the screwthreads does not exceed 0.020 in (AMM 27-50-141).	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270055	Flap synchronising system: <ul style="list-style-type: none"> • Check the cable tensions are correct (AMM 27-50-02). • With the flaps selected up, disconnect the operating link from one transmitter gearbox only. • Pressurize the hydraulic system. • Whilst restraining the disconnected lever arm, select the flaps down. • Make sure that the flaps start to move and then the system cuts out. • Depressurize the hydraulic system and connect the transmitter operating link. • Pressurize the hydraulic system and make sure that the flap operate correctly. 	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
270056	Airbrake control cables. Check the cable tensions are correct (AMM 27-61-02).	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270057	Flap control forward spring strut. Check break-out load with one end disconnected and using a spring balance is 21.5 to 24.5 lb. (AMM 27-50-02).	D4	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
270058	Flap control input-circuit. Do a static friction check (AMM 27-50-02).	D4	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
270059	Control cable pressure seals at rear pressure bulkhead. Inspect for wear.	E7	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
270060	Filter, auxiliary hydraulic system supply to flap control unit. Replace the filter element.	D10	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
270064	Flap vane inner. With the flaps lowered, inspect the left and right outer attachment bracket for cracking and for security of attachment to the flap leading edge.	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
270065	Flap screwjacks. Inspect the screw threads for satisfactory condition.	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270066	Aileron Control cable pressure seals. Inspect for wear.	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
270067	Aileron Control cable pressure seals. Inspect for wear.	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
27-COMPT	Flap Screwjack & Reduction Gearbox LH	O/H	4000 Ldgs	20-Dec-14	6,756.42	3,974	-	-	7974	2755.0	Cyl
27-COMPT	Flap Screwjack & Reduction Gearbox RH	O/H	4000 Ldgs	20-Aug-15	7,068.50	4,329	-	-	8329	3110.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Inner) RH FORWARD	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Outer) RH FORWARD	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Outer) RH AFT	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Inner) LH AFT	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Inner) LH FORWARD	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Outer) LH AFT	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Outer) LH FORWARD	Replace	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	9866	4647.0	Cyl
27-COMPT	Flap Synchro Cable Assy (Inner) RH AFT	O/H	5000 Ldgs	2-Jun-10	-	2,767	-	-	7767	2548.0	Cyl
27-COMPT	Flap Control Unit	O/H	6000 Ldgs	27/07/1997	-	-	-	-	6000	781.0	Cyl
27-COMPT	Flap Synchroniz Sys Receiver Unit	O/H	4500 Ldgs	20/8/2015	7,068.50	4,329	-	-	8829	3610.0	Cyl
27-COMPT	Elevator Trim Servo/Motor	O/H	5000 Flight Hrs	22/12/2015	7,194.90	4,517	-	12194.90	-	4363.8	Hrs
27-COMPT	Airbrake Jack Body LH	O/H	25000 Ldgs	27/07/1997	-	-	-	-	25000	19781.0	Cyl
27-COMPT	Airbrake Jack Body RH	O/H	25000 Ldgs	27/07/1997	-	-	-	-	25000	19781.0	Cyl
27-COMPT	Airbrake Jack LH	O/H	3000 Flight Hrs	10-Mar-11	5,775.80	-	-	8775.80	-	944.7	Hrs
27-COMPT	Airbrake Jack RH	O/H	3000 Flight Hrs	10-Mar-11	5,775.80	-	-	8775.80	-	944.7	Hrs
28	Fuel										
280001	Fuel Transfer System, Wing tanks. Check fuel flow. • Check for positive flow from the LH and the RH wing tank to the opposite wing tank. NOTE For positive flow note a 200lb increase in the fuel gauge indication	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
280002	Fuel valve control cables. Check cable tensions are correct (AMM 28-20-83).	C5	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
280003	LP valves. Do a function test for internal leakage (AMM 28-20-00).	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
280005	Suction pipe strainers. Make sure they are clean.	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
280006	Water transfer system pipe suction blocks. Make sure they are free from obstruction.	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
280007	Suction pipe strainers. Make sure they are clean.	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
280008	Water transfer system pipe suction blocks. Make sure they are free from obstruction.	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
280014	Applicable to Intertechnique fuel pumps only. Fuel system pumps: • Part No. 2070C01 Check the brush length (14 mm minimum). • Part No. 2070C11 Check the brush length (10 mm minimum).	C9	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
280015	Applicable to Intertechnique fuel pumps only. Fuel system pumps: • Part No. 2070C01 Check the brush length (14 mm minimum). • Part No. 2070C11 Check the brush length (10 mm minimum)	C10	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
29 Hydraulic Power											
290003	Suction filter. Replace the element (AMM 29-10-21).	B7	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
290004	Main hydraulic system. • At system shutdown check the pressure dissipation time on the main pressure accumulator (AMM 29-10-00, 501)	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
290005	Landing gear and flaps emergency lowering system. Do an operational test (AMM 29-20-00).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
290006	Hydraulic tank pressurization system. Drain any accumulation of liquid from the drain/ground air charge connection.	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
290007	Flaps, airbrakes, landing gear, nosewheel steering, nose leg self-centring. Do a test of the main system with a hydraulic test rig (AMM 29-10-00).	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
290008	Emergency brake system leak rate. Check the rate that the pressure decreases (leakage rate) in the accumulator (AMM 29-10-00).	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
290009	Auxiliary reservoir low-level warning system. Do a function test (AMM 29-30-00). B12 NOTE Do this task in conjunction with Task # 290005.	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
290010	Hydraulic reservoir pressurization system. Do a functional test of the non-return valve (AMM 29-10-13). C12	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
290011	Full-flow relief valve. Clean filter element (AMM 29-10-71). D7	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
290012	System overheat temperature switch. • Inspect. • Check operating datums (AMM 29-30-00).	D7	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
29-COMPT	Full Flow Relief Valve #1	Replace	4500 Ldgs	20-Aug-15	7,068.50	4,329	-	-	8829	3610.0	Cyl
29-COMPT	Full Flow Relief Valve #2	Replace	4500 Ldgs	20-Aug-15	7,068.50	4,329	-	-	8829	3610.0	Cyl
29-COMPT	Pressure Maintaining Valve	Replace	5000 Ldgs	11-Aug-17	7,709.80	5,131	-	-	10131	4912.0	Cyl
29-COMPT	Emergency Brake Acc Low Pres Warning Switch	Replace	5000 Flight Hrs	8-Apr-09	5,155.70	-	-	10155.70	-	2324.6	Hrs
30 Ice and Rain Protection											
300001	Rotary ice detector. Do a function test (AMM 30-80-00).	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
300002	Windshield electrical de-icing system. Do an operational test of the windscreen overheat (AMM 30-41-00).	B1	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
300003	Airframe de-icing system pressure filter. • Replace the filter element (AMM 30-10-52).	B6	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
300004	Pitot head and static plate heaters. Do a function test (AMM 30-35-15).	C1	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
300005	Airframe de-icing pump suction-filter. Replace the filter element (AMM 30-10-43).	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
31 Indicating / Recording systems											
310001	Flight data recorder. Check for correct playback operation (data dump) and for quality of the recording. NOTE: Do the check at this inspection or at a time agreed necessary by the local airworthiness authority.	E1	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	9309.80	-	-121.0	Days
31-COMPT	Flight Data Recorder readout	Readout	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
32	Landing Gear										
320001	Braking systems. • Do a test of the emergency and parking brake system operation (AMM 32-40-00, 501).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320002	Maxaret Units. • Do a function test (AMM 32-40-101,501).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320003	Nose gear leg assembly: • Make sure oil level and gas pressures are correct (AMM 32-20-12). • Check the leg extension (AMM 12-32-00).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320004	Nose gear leg lock and change-over microswitches operating mechanism. Make sure of freedom of operation.	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320005	Nosewheel steering jack: • Inspect for satisfactory condition. NOTE Pay particular attention to scores and the cleanliness of the jack ram.	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320006	Nose gear retraction and steering system spring struts. Inspect.	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320007	Nosewheel steering ON/OFF valve. Inspect.	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320008	Nose gear drag stay assembly. Check clearances (AMM 32-20-21).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320009	Nosewheel steering jack selector valve: • Detail inspect the steering jack (AMM 32-50-11).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320010	Main gear leg assemblies • Make sure oil level and gas pressures are correct (AMM 32-10-12). • Check the leg extension (AMM 12-32-00)	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320011	Main landing gear side-stays. • Inspect the center joint nut for distortion (AMM 32-10-25). • Check the side-stay clearances (AMM 32-10-25).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320012	Main landing gear torque links. Examine the torque link pivots for wear (AMM 32-10-12, page block 601).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320013	Main landing gear doors. • Inspect the hinge bracket assemblies for security and tightness of the nuts. • Inspect jack attachment brackets and adjacent structure for cracks. • Inspect for signs of movement of the forward hinge brackets and bolts at their attachment to the wing structure by attempting to rock the doors in the longitudinal and transverse directions. NOTE If the brackets are found to be loose, inspect the attachment bolt holes and bolts for wear and corrosion.	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320014	Main landing gear door uplock spring struts. • Inspect. • Clean the sliding surfaces. • Make sure of correct operation	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
320015	Reversing valves and operating mechanisms: <ul style="list-style-type: none"> • Disconnect from the side stay. • Inspect for satisfactory condition. • Make sure the mechanisms are free to move correctly. • Connect to the side stay. 	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320016	Steering system cables and chains. Make sure cable tensions are correct (AMM 32-50-00).	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320017	Nose landing gear retraction spring strut. Check the break-out load (AMM 32-20-21).	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320018	Nosewheel steering system spring strut break-out load. <ul style="list-style-type: none"> • Check the break-out load (CMM 32-50-00). • If the load is excessive, dismantle the spring strut, clean, lubricate and reassemble (CMM 32-50-00). 	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320019	Steering system filter element. Remove, inspect and clean filter element (AMM 32-30-22).	D12	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
320020	Filters, hydraulic supply (4 off). Clean the filter elements (AMM 32-30-22).	D12	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
320021	Restrictor valves. Clean the micro filters in the pipe connections (AMM 32-30-01).	D12	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
320022	Brake master cylinders. Check fluid level and replenish, as necessary (AMM 32-40-00).	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320023	Brake hoses to master cylinders. Make sure of an adequate clearance between the hoses and the floor over the full range of rudder pedal travel.	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320024	Axle/Maxaret/Modulator/Plunger Tube <ul style="list-style-type: none"> • Maxaret and modulator unit. Remove and inspect(AMM 32-40-91 and AMM 32-40-101). • While the Maxarets and Modulator units are removed, detail inspect the internal bores of the axles. • Inspect the Plunger Tube to Axle Bore for signs of corrosion. • Reinstall Maxarets and Modulator units(AMM 32-40-91 and AMM 32-40-101). • Do a function test (AMM 32-40-101, ADJUSTMENT/TEST) 	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320027	Nose-wheel steering on/off-valve: <ul style="list-style-type: none"> • Remove the plunger stop cap. • Visually inspect the plunger and spring for corrosion, clean and lubricate all surface areas as required using Grease (Ref. AMM 125/H-20, 20-95-11, Item 03 • Install the stop cap. 	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
320028	8 years, then every 4 years. Main landing gear fairing strut. <ul style="list-style-type: none"> • Comply with instructions in NTM 32-10-103 	Structural	96 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
320028	8 years, then every 4 years. Main landing gear fairing strut. <ul style="list-style-type: none"> • Comply with instructions in NTM 32-10-103 	Structural	96 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
320033	Main gear retraction actuator (AMM 32-30-27). NOTE To do this inspection, disconnect the cylinder head end of the actuator. Do not disconnect the hydraulic lines. • Detail inspect the actuator attachment lugs	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320034	Plunger Tube/Axle Joint • Inspect the flexible seal at the plunger tube to axle joint. • Make sure the seal is not loose or cracked. • Rectify as necessary (AMM Chapter 32, MAIN LEG ASSEMBLY)	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
320035	Main Leg Assembly Retraction Lever • Inspect for excessive movement. • Inspect for signs of corrosion	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320036	Sidestay Assembly Lock Mechanism Link • Inspect for signs of wear.	C12	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
320037	Nose leg assembly torque links. Examine the torque link pivots for wear (AMM 32-20-12, 601).	B12	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
32-COMPT	Main Gear Leg Assembly LH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Leg Assembly LH	O/H	144 Months	30-Nov-09	5,345.40	2,587	27-Nov-21	-	-	1083.0	Days
32-COMPT	Main Gear Leg Assembly RH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Leg Assembly RH	O/H	144 Months	30-Nov-09	5,345.40	2,587	27-Nov-21	-	-	1083.0	Days
32-COMPT	Main Gear Side Stay LH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Side Stay LH	O/H	144 Months	30-Nov-09	5,345.40	2,587	27-Nov-21	-	-	1083.0	Days
32-COMPT	Main Gear Side Stay RH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Side Stay RH	O/H	144 Months	30-Nov-09	5,345.40	2,587	27-Nov-21	-	-	1083.0	Days
32-COMPT	Main Gear Torque Link Pin LH UPPER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Torque Link Pin RH UPPER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Sidestay Upper Arm LH	Replace	10000 Ldgs	27/07/1997	-	-	-	-	10000	4781.0	Cyl
32-COMPT	Main Gear Sidestay Upper Arm RH	Replace	10000 Ldgs	27/07/1997	-	-	-	-	10000	4781.0	Cyl
32-COMPT	Main Gear Sidestay Lower Arm LH	Replace	12000 Ldgs	27/07/1997	-	-	-	-	12000	6781.0	Cyl
32-COMPT	Main Gear Sidestay Lower Arm RH	Replace	12000 Ldgs	27/07/1997	-	-	-	-	12000	6781.0	Cyl
32-COMPT	Main Gear Plunger Tube LH	Replace	17000 Ldgs	27/07/1997	-	-	-	-	17000	11781.0	Cyl
32-COMPT	Main Gear Plunger Tube RH	Replace	17000 Ldgs	27/07/1997	-	-	-	-	17000	11781.0	Cyl
32-COMPT	Main Gear Upper Casing LH	Replace	24000 Ldgs	27/07/1997	-	-	-	-	24000	18781.0	Cyl
32-COMPT	Main Gear Upper Casing RH	Replace	24000 Ldgs	27/07/1997	-	-	-	-	24000	18781.0	Cyl
32-COMPT	Nose Gear Axle Barrel	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Nose Steering Jack Pick Up	Replace	18000 Ldgs	27/07/1997	-	-	-	-	18000	12781.0	Cyl
32-COMPT	Nose Gear Torque Link UPPER	Replace	32000 Ldgs	27/07/1997	-	-	-	-	32000	26781.0	Cyl
32-COMPT	Nose Gear Drag Stay	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Nose Gear Drag Stay	O/H	144 Months	30-Nov-09	5,345.40	2,587	27-Nov-21	-	-	1083.0	Days
32-COMPT	Nose Steering Main Member UPPER	Replace	35000 Ldgs	27/07/1997	-	-	-	-	35000	29781.0	Cyl
32-COMPT	Nose Gear Connecting Rod	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Nose Steering Sleeve	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Nose Gear Torque Link Pickup Sleeve	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Nose Gear Drag Stay Bridge Casting	Replace	32000 Ldgs	27/07/1997	-	-	-	-	32000	26781.0	Cyl
32-COMPT	Nose Gear Leg Assembly	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Nose Gear Leg Assembly	O/H	144 Months	30-Nov-09	5,345.40	2,587	27-Nov-21	-	-	1083.0	Days
32-COMPT	Main Gear Jack Pivot Bolt LH	Replace	30000 Ldgs	27/07/1997	-	-	-	-	30000	24781.0	Cyl
32-COMPT	Main Gear Jack Pivot Bolt RH	Replace	30000 Ldgs	27/07/1997	-	-	-	-	30000	24781.0	Cyl
32-COMPT	Main Landing Gear Jack Distance Pin LH	Replace	24000 Ldgs	27/07/1997	-	-	-	-	24000	18781.0	Cyl
32-COMPT	Main Landing Gear Jack Distance Pin RH	Replace	24000 Ldgs	27/07/1997	-	-	-	-	24000	18781.0	Cyl
32-COMPT	Main Gear Retraction Arm Pin LH	Replace	32000 Ldgs	27/07/1997	-	-	-	-	32000	26781.0	Cyl
32-COMPT	Main Gear Retraction Arm Pin RH	Replace	32000 Ldgs	27/07/1997	-	-	-	-	32000	26781.0	Cyl

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
32-COMPT	Main Gear Retraction Arm LH	Replace	12000 Ldgs	27/07/1997	-	-	-	-	12000	6781.0	Cyl
32-COMPT	Main Gear Retraction Arm RH	Replace	12000 Ldgs	27/07/1997	-	-	-	-	12000	6781.0	Cyl
32-COMPT	Nose Gear Torque Link LOWER	Replace	32000 Ldgs	27/07/1997	-	-	-	-	32000	26781.0	Cyl
32-COMPT	Nose Gear Jack Lower Pivot Pin	Replace	22000 Ldgs	27/07/1997	-	-	-	-	22000	16781.0	Cyl
32-COMPT	Nose Gear Main Casing Bolt	Replace	32000 Ldgs	27/07/1997	-	-	-	-	32000	26781.0	Cyl
32-COMPT	Nose Steering Pin	Replace	25000 Ldgs	27/07/1997	-	-	-	-	25000	19781.0	Cyl
32-COMPT	Nose Landing Gear Pivot Pin	Replace	16000 Ldgs	27/07/1997	-	-	-	-	16000	10781.0	Cyl
32-COMPT	Nose Gear Plunger Tube	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Nose Gear Radius Rod Pick-Up Bolts	Replace	10000 Ldgs	27/07/1997	-	-	-	-	10000	4781.0	Cyl
32-COMPT	Nose Gear Pick-Up Trunnion Bolts	Replace	10000 Ldgs	27/07/1997	-	-	-	-	10000	4781.0	Cyl
32-COMPT	Brake Emer Reducing Valve Knife Edg LH	Replace	4500 Ldgs	22-Dec-15	7,194.90	4,517	-	-	9017	3798.0	Cyl
32-COMPT	Brake Maxaret LH INBD	O/H	3000 Ldgs	17-Jun-11	-	3,171	-	-	6171	952.0	Cyl
32-COMPT	Brake Maxaret RH OTBD	O/H	3000 Ldgs	3-Jan-11	-	2,998	-	-	5998	779.0	Cyl
32-COMPT	Brake Maxaret LH OTBD	O/H	3000 Ldgs	3-Jan-11	-	2,998	-	-	5998	779.0	Cyl
32-COMPT	Brake Maxaret RH INBD	O/H	3000 Ldgs	3-Jan-11	-	2,998	-	-	5998	779.0	Cyl
32-COMPT	Brake Control Valve Knife Edge RH OUTER	Replace	4500 Ldgs	14/08/2003	2,586.40	1,275	-	-	5775	556.0	Cyl
32-COMPT	Brake Emer Reducing Valve Knife Edg RH	Replace	4500 Ldgs	22-Dec-15	7,194.90	4,517	-	-	9017	3798.0	Cyl
32-COMPT	Brake Control Valve Knife Edge LH OUTER	Replace	4500 Ldgs	14/08/2003	2,586.40	1,275	-	-	5775	556.0	Cyl
32-COMPT	Brake Control Valve Knife Edge INNER	Replace	4500 Ldgs	14/08/2003	2,586.40	1,275	-	-	5775	556.0	Cyl
32-COMPT	Nose Gear Steering Jack Eye End	Replace	14400 Ldgs	27/07/1997	-	-	-	-	14400	9181.0	Cyl
32-COMPT	Main Gear Jack Cylinder Head LH	Replace	12000 Ldgs	27/07/1997	-	-	-	-	12000	6781.0	Cyl
32-COMPT	Main Gear Jack Cylinder Head LH	Replace	168 Months	24-Jul-14	6,626.40	3,798	20-Jul-28	-	-	3510.0	Days
32-COMPT	Main Gear Jack Cylinder Head RH	Replace	12000 Ldgs	27/07/1997	-	-	-	-	12000	6781.0	Cyl
32-COMPT	Main Gear Jack Cylinder Head RH	Replace	168 Months	24-Jul-14	6,626.40	3,798	20-Jul-28	-	-	3510.0	Days
32-COMPT	Main Gear Jack Eye End LH	Replace	10000 Ldgs	27/07/1997	-	-	-	-	10000	4781.0	Cyl
32-COMPT	Main Gear Jack Eye End RH	Replace	10000 Ldgs	27/07/1997	-	-	-	-	10000	4781.0	Cyl
32-COMPT	Nose Gear Retract Jack	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Door Jack RH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Door Jack LH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Retract Jack LH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Retract Jack RH	O/H	5000 Ldgs	30-Nov-09	5,345.40	2,587	-	-	7587	2368.0	Cyl
32-COMPT	Main Gear Reversing Valve LH	O/H	5000 Ldgs	7-Oct-08	5,048.54	2,417	-	-	7417	2198.0	Cyl
32-COMPT	Main Gear Reversing Valve LH	O/H	144 Months	7-Oct-08	5,048.54	2,417	4-Oct-20	-	-	664.0	Days
32-COMPT	Main Gear Reversing Valve RH	O/H	5000 Ldgs	2-Jun-10	5,543.50	2,767	-	-	7767	2548.0	Cyl
32-COMPT	Main Gear Reversing Valve RH	O/H	144 Months	2-Jun-10	5,543.50	2,767	30-May-22	-	-	1267.0	Days
32-COMPT	Brake Emergency Sys Pressure Switch LH	O/H	5000 Flight Hrs	28/02/2008	4,735.10	2,268	-	9735.10	-	1904.0	Hrs
32-COMPT	Brake Emergency Sys Pressure Switch RH	O/H	5000 Flight Hrs	28/02/2008	4,735.10	2,268	-	9735.10	-	1904.0	Hrs
32-COMPT	Drag Strut Pick Up	Replace	13900 Ldgs	27/07/1997	-	-	-	-	13900	8681.0	Cyl
32-COMPT	Main Gear Torque Link LH LOWER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Torque Link RH LOWER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Torque Link LH UPPER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Torque Link RH UPPER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Torque Link Pin LH LOWER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Main Gear Torque Link Pin RH LOWER	Replace	20000 Ldgs	27/07/1997	-	-	-	-	20000	14781.0	Cyl
32-COMPT	Nose Steering Main Member LOWER	Replace	35000 Ldgs	27/07/1997	-	-	-	-	35000	29781.0	Cyl
32-AD	To Prevent The Failure Of The Knife Edge Of The Brake Control Valve, P/N Ac.61520 & The Loss Of Braking On One Side	AD	4500 Ldgs	9-Jun-16	7,438.80	4,775	-	-	9275	4156.0	Cyl
32-AD	To Prevent Possible Failure Of The Knife Edges Of The Emerg Brake Reducing Valve P/N Ac.61516 Which Results In Loss Brake	AD	4500 Ldgs	9-Jun-16	7,438.80	4,775	-	-	9275	4256.0	Cyl
32-AD	To Ensure Continued Structural Integrity Of The Main Landing Gear Sidestay Upper Arm Assembly	AD	12000 Ldg	27-Jan-93	-	-	-	-	12000	6781.0	Cyl
32-AD	LH Dunlop Brake Control Valve, Failure of Knife Edge	SB	4500 Ldgs	9-Jun-16	7,438.80	4,775	-	-	9275	4156.0	Cyl
32-AD	Dunlop Emergency Brake Reducing Valve, Failure of Knife Edge	SB	4500 Ldgs	22-Dec-15	7,194.90	4,517	-	-	9017	3798.0	Cyl
32-AD	MLG Side Stay Upper Arm, Introduction of Ultimate Life	SB	12000Cy	21-Mar-88	-	-	-	-	12000	6781.0	Cyl
32-AD	MLG Side Stay Upper Arm, Introduction of Ultimate Life	SB	12000Cy	21-Mar-88	-	-	-	-	12000	6781.0	Cyl
32-AD	NLG Pivot Pin Crack Test	SB	4000 Ldgs	20-Dec-14	6,756.42	3,974	-	-	7974	2755.0	Cyl

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
32-AD	NLG Assy	SB	12 Years	3-Nov-01	-	-	5-Oct-21	-	-	37554.0	Days
32-AD	NLG Assy	SB	5000 Ldgs	3-Nov-01	6,038.10	3,340	-	-	8340	3121.0	Cyl
53-AD	Inspection of NLG Bay Sidewall for cracks	SB	4000 Ldgs	20-Dec-14	6,756.42	3,974	-	-	7974	2755.0	Cyl
33	Lights										
330003	Anti-Collision light system that has a strobe type device in any position. <ul style="list-style-type: none"> Clean the strobe light lenses internal and external. Clean the wing-tip lens covers internal and external. Check the strobe light flash rate (Not less than 40 and no more than 100 cycles per minute). NOTE Refer to the Component Requirements section of this manual for the manufacturer's recommended period for replacement and strobe light intensity test procedure.	B1	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
33-COMPT	Strobe Light Assembly Tail	OOP	1500 Flight Hrs	2-Feb-14	6,475.70	3,622	-	7975.70	-	144.6	Hrs
33-COMPT	Strobe Light Assembly LH	OOP	1500 Flight Hrs	19-Aug-16	7,523.40	4,866	-	9023.40	-	1192.3	Hrs
33-COMPT	Strobe Light Assembly RH	OOP	1500 Flight Hrs	19-Aug-16	7,523.40	4,866	-	9023.40	-	1192.3	Hrs
33-COMPT	Emergency Light Battery Pack AFT	OOP	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
33-COMPT	Emergency Light Battery Pack FWD	OOP	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
34	Navigation										
340002	Pitot/static drain traps. <ul style="list-style-type: none"> Pre SB.34-3282 (Mod. 25F889B) or Pre Mod. 25F889A. Drain, do a leak test (AMM 34-11-19). Post SB.34-3282 (Mod. 25F889B) or Post Mod. 25F889A. Drain (AMM 34-11-19). NOTE: Do this task in conjunction with FAR pitot/static system checks (Task 340006). Autopilot aileron servo unit and mount	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
340003	Autopilot aileron servo unit and mount. <ul style="list-style-type: none"> Inspect, particularly for cable and capstan wear, contamination, cable spool-off angle and security of the servo unit and mount. With the autopilot disengaged operate the control system throughout the entire range and make sure there is no unusual noise, binding, backlash or signs of other mechanical irregularities from the servo mount 	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
340004	Autopilot servo units, rudder and elevator. <ul style="list-style-type: none"> Inspect, particularly for cable and capstan wear, contamination, cable spool-off angle and security of the servo unit and mount. With the autopilot disengaged operate the control system throughout the entire range and make sure there is no unusual noise, binding, backlash or signs of other mechanical irregularities from the servo mount 	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
340005	Autopilot elevator trim servo unit and mount. <ul style="list-style-type: none"> Inspect, particularly for cable and capstan wear, contamination, cable spool-off angle and security of the servo unit and mount With the autopilot disengaged operate the control system throughout the entire range and make sure there is no unusual noise, binding, backlash or signs of other mechanical irregularities from the servo mount. 	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
340006	Pitot/static and stall warning systems. Do the following tests (AMM 34-11-19, 501): <ul style="list-style-type: none"> Pressure Test Complete Pitot System. Suction Test Complete Static System. Pressure Test Stall Vent System. NOTE Do this task in accordance with FAR requirements	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
340007	Pitot system isolation valves. <ul style="list-style-type: none"> Make sure of full and free operation. NOTE: Do this task in conjunction with FAR pitot/static system checks (Task 340006).	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
340008	Mach trim system. Do a test of the Mach trim system: <ul style="list-style-type: none"> 800XP airplanes (AMM 34-24-00, 501) 	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
340010	Autotrim command isolation relay and autotrim switching relay: NOTE This task is ONLY applicable to airplanes equipped with an FCC-86 Flight Control Computer. <ul style="list-style-type: none"> Do checks as detailed, Ref. AMM 27-30-00, 501, (Autotrim command isolation relay and autotrim switching relay checks). 	D1	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
340011	Mach trim fail annunciation relays. <ul style="list-style-type: none"> Open the MACH TRIM PWR or MACH TRIM MON circuit breaker. Make sure that the MACH TRIM FAIL annunciator comes on. Close the MACH TRIM PWR or the MACH TRIM MON circuit breaker. Make sure that the MACH TRIM FAIL annunciator goes off. Every 36 months, or at a time period agreed with the local regulatory authority.	D1	3200 Flight Hrs or 36 Months (Which ever is earlier)	20-Aug-15	7,068.50	4,329	19-Aug-18	10268.50	-	-113.0	Days
340012	Compass systems: <ul style="list-style-type: none"> Do a check compass swing of the compass systems at four headings 90 degrees apart. Compare deviations, if any are found, with the existing deviation card. If deviations are found then do a compass swing: 800XP airplanes (AMM 34-23-61, 501).	OOP	36 Months	11-Aug-17	7,709.80	5,131	10-Aug-20	-	-	609.0	Days
34-COMPT	Altimeter STANDBY	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Altimeter LH	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
34-COMPT	Altimeter RH	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Air Data Computer #1	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Air Data Computer #2	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Transponder #1	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Transponder #2	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Static Air Leak CheckLH	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Static Air Leak CheckRH	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-COMPT	Static Air Leak Check S-BY	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-RVSM	RVSM Check,	OOP	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
34-SB	Air Data Computer, Increase Accuracy	SB	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
35	Oxygen										
350001	Oxygen system master supply valve: <ul style="list-style-type: none"> With the oxygen system fully charged and the supply master valve closed, exhaust system downstream of the valve until the contents gauge indicates zero. After 15 minutes has elapsed, make sure that there is no increase in the contents gauge indication which would show leakage past the closed valve. 	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
350002	Automatic shut-off valve. Do a functional test (AMM 35-10-141).	D2	3200 Flight Hrs	20-Aug-15	7,068.50	4,329	-	10268.50	-	2437.4	Hrs
350004	Passenger oxygen system. <ul style="list-style-type: none"> Test automatic operation of baromatic valve (AMM 35-06-00, 501). Test for flow from each mask by reference to operation of the visual indicator when mask is pulled down to the 'full hang' position: 800XP and 800XP Pro Line 21 airplanes (MMA 35-06-00, 201). Inspect drop-out mask units when deployed during system test: 800XP and 800XP Pro Line 21 airplanes (MMA 35-20-05, 301). Stow drop-out mask units: 800XP and 800XP Pro Line 21 airplanes (MMA 35-20-05, 301). 	B2	800 Flight Hrs	11-Aug-17	7,709.80	5,131	-	8509.80	-	678.7	Hrs
350005	Baromatic control valve. <ul style="list-style-type: none"> With passenger supply isolation valve closed, exercise control valve mechanism by operating the emergency/reset knob several times. Leave the knob at the reset position. 	B2	800 Flight Hrs	11-Aug-17	7,709.80	5,131	-	8509.80	-	678.7	Hrs
350006	Crew masks, smoke set mask and hoses. Inspect.	B4	800 Flight Hrs	11-Aug-17	7,709.80	5,131	-	8509.80	-	678.7	Hrs
350007	Portable breathing equipment: <ul style="list-style-type: none"> Inspect: 800XP airplanes (AMM 35-35-10). OR 750, 800XP, 800XP Proline 21 and 900XP (AMM 35-35-15). Check the expiry dates. 	E4	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
350008	Oxygen cylinder. <ul style="list-style-type: none"> Inspect cylinder mount clamps for security 	C7	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
35-COMPT	Oxygen Container (Portable)	OOP	60 Months	2-Feb-14	6,475.70	3,622	1-Feb-19	-	-	53.0	Days
35-COMPT	Oxygen Mask PAX #1	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #10	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #11	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #12	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #2	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #3	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #4	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
35-COMPT	Oxygen Mask PAX #5	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #6	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #7	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #8	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask PAX #9	OOP	72 Months	11-Aug-17	7,709.80	5,131	10-Aug-23	-	-	1704.0	Days
35-COMPT	Oxygen Mask COPILOT	OOP	72 Months	22-Dec-15	7,194.90	4,517	31-Aug-21	-	-	995.0	Days
35-COMPT	Oxygen Mask PILOT	OOP	72 Months	21-Dec-12	6,237.40	-	20-Dec-18	-	-	10.0	Days
35-COMPT	Oxygen Container #3	OOP	60 Months	20-Dec-14	6,756.42	3,974	19-Dec-19	-	-	374.0	Days
35-COMPT	Oxygen Container #1	OOP	60 Months	20-Dec-14	6,756.42	3,974	19-Dec-19	-	-	374.0	Days
35-COMPT	Oxygen Container #2	OOP	60 Months	20-Dec-14	6,756.42	3,974	19-Dec-19	-	-	374.0	Days
35-COMPT	Oxygen Container #4	OOP	60 Months	20-Dec-14	6,756.42	3,974	19-Dec-19	-	-	374.0	Days
35-COMPT	Protective Breathing Equipment	OOP	120 Months	4-Nov-11	6,068.40	-	1-Nov-21	-	-	1057.0	Days
38	Water/Waste										
380001	<p>Water systems:</p> <ul style="list-style-type: none"> Inspect the toilet and galley water systems. Make sure the flow of water is correct through the normal and overflow drains and that there are no leaks. <p>If you find signs of fluid leakage:</p> <ul style="list-style-type: none"> Inspect the aileron-cable fuselage pressure-seals and the adjacent structure in the areas to the rear of frame 16. Do this to make sure the fluid leakage has drained correctly. Make sure the drain holes and drain valves are free from blockages and function correctly. Do this from the inside and the outside of the fuselage. Remove all fluid remaining in the area. Clean the areas where leakage has occurred. Lubricate the aileron cables if necessary (AMM 12-20-25). Replenish the toilet and galley water systems, if required: 800XP airplanes (AMM 12-38-00 or MMA Chapter 38 as applicable). 	B6	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
38-AD	Inspect Lavatory Waste Receptacle: To Prevent Possible Fires That Could Result From Smoking Materials Dropped In Recept Narco Model Elt-10 Emergency Locator	AD	1000 Flight Hrs	19-Aug-16	7,523.40	4,866	-	8523.40	-	692.3	Hrs
49	Auxilliary Power Unit										
490001	<p>Perform in conjunction with Task 490003.</p> <p>PATS APU GTCP36-150W.</p> <ul style="list-style-type: none"> Visually inspect exhaust for cracks or deformation. Visually inspect for loose or frayed wires. Inspect for signs of corrosion 	OOP	1000 APU Hrs or 12 Months (Which ever is earlier)	21-Dec-16	7,523.40	4,866	21-Dec-17	8523.40	-	-354.0	Days
490002	<p>Every 750 APU running hours.</p> <p>PATS GTCP36-150W APU.</p> <p>Inspect Igniter plug (CMM 49-40-33):</p> <ul style="list-style-type: none"> Remove, clean, inspect and install 	OOP	750 APU Hrs	21-Dec-16	7,523.40	4,866	-	8273.40	-	3388.4	APU Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
490003	<p>NOTE: Choose one of the following options. Do either Oil and Filter change at every 12 months or 1000 flying hours OR do a S.O.A.P. check at 750 APU running hours/12 months. CAUTION: Do not mix options.</p> <p>Option 1. Every 12 months or 1000 flying hours (whichever comes first). PATS GTCP36-150W APU. Oil and filter change, and magnetic drain plug inspection. Lubricating oil system: • Drain and refill (refer to servicing). NOTE: Routine SOAP inspection is not required.</p> <p>Oil Filter Element: • Remove, inspect and replace as necessary (CMM 49-90-47). Magnetic Drain Plug: • Remove, clean, inspect and install magnetic element of drain plug (CMM 49-90-46).</p> <p>Option 2. 750 APU running hours or 12 months (whichever comes first). Oil and Filter Element: • Do a S.O.A.P. check (in accordance with Honeywell procedures). Magnetic Drain Plug: • Remove, clean, inspect and install magnetic element of drain plug (CMM 49-90-46).</p>	OOP	1000 APU Hrs or 12 Months (Which ever is earlier)	21-Dec-16	7,523.40	4,866	21-Dec-17	8523.40	-	-354.0	Days
490004	<p>Every 1000 APU running hours. PATS GTCP36-150W APU. Fuel filter element: • Remove and replace (CMM 49-30-52).</p>	OOP	1000 APU Hrs	18-Aug-15	4,494.00	-	-	5494.00	-	609.0	APU Hrs
490005	<p>Every 4500 APU running hours. PATS GTCP36-150W APU. Fuel Nozzle Assemblies: Do a fuel flow rate check (CMM 49-20-00, Page Block 601, paragraph 5, Fuel Nozzle Assembly Flow Rate Check).</p>	OOP	4500 APU Hrs	18-Aug-15	4,494.00	-	-	8994.00	-	4109.0	APU Hrs
490006	<p>Every 4500 APU running hours. PATS GTCP36-150W APU. Hot Section Inspection. Do a detailed Hot Section Inspection on the components listed (CMM 49-20-00, Page Block 601, paragraph (4), Detailed Hot Section Inspection):</p> <ul style="list-style-type: none"> • Turbine Rotor • Deswirl Deflector • Turbine Nozzle • Airflow Deflector • Turbine Housing Assembly • Combustion Chamber • Labyrinth Seal • Turbine Wheel Deflector • Retainer Halves • Containment Ring <p>Every 4500 APU running hours. PATS GTCP36-150W APU. Compressor Rotor: • Visually check compressor rotor for damage by observation through the compressor screen openings. NOTE: No damage allowed.</p>	OOP	4500 APU Hrs	18-Aug-15	4,494.00	-	-	8994.00	-	4109.0	APU Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
490007	Every 4500 APU running hours. PATS GTCP36-150W APU. Compressor Rotor: • Visually check compressor rotor for damage by observation through the compressor screen openings. NOTE: No damage allowed.	OOP	4500 APUHrs	18-Aug-15	4,494.00	-	-	8994.00	-	4109.0	APU Hrs
49-COMPT	Apu Starter/Generator	OOP	600 APU Hrs	21-Dec-16	7,523.40	4,866	-	8123.40	-	3238.4	APU Hrs
49-COMPT	Apu Starter/Generator	OOP	1000 APU Hrs	21-Dec-16	7,523.40	4,866	-	8523.40	-	3638.4	APU Hrs
49-COMPT	Apu Turbine Rotor	OOP	10000 Ldgs	2/10/1997	-	-	-	-	10000	4781.0	APU Cyl
49-COMPT	Apu Turbine Rotor	OOP	10000 APU Hrs	2/10/1997	-	-	-	10000.00	-	5115.0	APU Hrs
49-COMPT	Apu Compressor Rotor	OOP	99999 APU Hrs	2/10/1997	-	-	-	99999.00	-	95114.0	APU Hrs
49-COMPT	Apu Compressor Rotor	OOP	99999 Ldgs	2/10/1997	-	-	-	-	99999	94780.0	APU Cyl
51	Standards Practices and Structures										
510001	Every 5000 landings or 12 years, whichever occurs first. Nose landing gear bay sidewalls. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-111.	Structural	5000 Landing or 144 Months (Which ever is earlier)	30-Nov-09	5,345.40	2,587	27-Nov-21	-	7587	1083.0	Days
510002	5000 flights then every 1000 flights. Front pressure bulkhead. Access: Sound-proofing blankets removed. • Detail inspect the upper sidewall beams and the rudder pedal clearance pressings for cracks (SRM 53-10-11), Front Pressure Bulkhead Pre Mod. 253090C or Front Pressure Bulkhead Mod. 253090C). • Pay special attention to the area in the vicinity of the joint with the canopy center post.	Structural	5000 Ldgs	19-Aug-16	7,523.40	4,866	-	-	5866	647.0	Cyl
510003	12 years, then every 4 years. Flight compartment skin and window structure above windshields. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-105A. Ref 53-19-105	Structural	144 Months	24-Jul-14	6,626.40	3,798	23-Jul-18	-	-	-140.0	Days
510004	Every 20 000 flights. Upper canopy rail at intersection with the canopy posts. Access: Trim and insulation removed. • Detail inspect.	Structural	20000 Ldgs	27-Jul-97	-	-	-	-	20000	14781.0	Cyl
510005	Between 15 to 18 years, then every 8 years. Center fuselage structure below the floor, nose landing gear bay to Frame 13 and Frames 16 to 19. Access: Refer to relevant NTM Techniques. • Comply with instructions in • NTM 53-10-108 • NTM 53-10-107 • NTM 53-10-108B • NTM 53-10-109.	Structural	216 Months	18-Aug-15	7,067.50	4,326	13-Aug-33	-	-	5360.0	Days
510006	Between 15 to 18 years and then every 8 years. Structure below escape hatch, Frames 13 to 14. Access: Escape hatch removed, leaded vinyl sleeve peeled back from lower corners of the escape hatch surround to expose the apertures in the surround. • Detail inspect Frames 13 and 14, as far as is possible, for signs of corrosion using a borescope probe.	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
510007	Between 15 to 18 years and then every 8 years. Center fuselage structure below the floor, frames 8 to 17. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-106A.	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
510008	Between 15 to 18 years and then every 8 years. Center fuselage structure below the floor, Frames 13 to 16. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-108A. NOTE: In certain instances it may be necessary to lower the wing or separate the wing from the fuselage to insert the film packs. In order to determine the conditions of individual airplanes a test may be done by attempting to pass a strip of vinyl carrier, to the dimensions given in the relevant NTM, between the fuselage lower skin and the wing top surface.	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
510009	8 years and then every 4 years. Entry door surround. Access: Panel F66A, F66B, F66C removed. • Detail inspect the structure at the lower edge of the surround for signs of corrosion using a borescope probe inserted through the lightening holes in the intercostal	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
510010	Between the 15th and 18th year of aircraft operation and then every 8 years. Center fuselage structure, above the floor. Access: Refer to relevant NTM Techniques. NOTE: The most economical and practical time to do this inspection is when the quick trim is removed for refurbishing or other reasons. NOTE: When installing local furnishings, make sure a minimum clearance of 1.0 inch between the motor of the fan operated venturi and local furnishings is kept. • Inspect the skin, as far as is possible, for cracks, signs of corrosion and deterioration of the protective treatment. • Comply with instructions in: • NTM 53-10-101. • NTM 53-10-102. • NTM 53-10-103A. • NTM 53-10-104. • NTM 53-10-106. NOTE: Should this inspection be done prior to the 15 years after manufacture point, then please inform: Service Engineering Manager, Dept. 211-B91 Customer Support, Hawker Beechcraft Corporation, P.O. Box 85, Wichita, Kansas, 67201-0085, USA.	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
52	Doors	0									
520002	Baggage pannier loading door/upper hatch warning system (iffitted) OR External baggage unit (EBU) (750 airplanes ONLY). • Test for correct operation (AMM 52-30-05). NOTE: This test is only applicable to airplanes with a baggage pannier in place of the ventral fuel tank OR 750 airplanes with external baggage unit.	C1	1600 Flight Hrs	4-Nov-11	6,068.40	-	-	7668.40	-	N/A	N/A

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
520003	<p>Baggage pannier door (if fitted) OR External baggage unit (EBU) (750 airplanes ONLY) . Inspect (AMM 52-30-05):</p> <ul style="list-style-type: none"> • The door seal for damage, distortion and maintained shape profile. • Seal for adhesion to the door lip. • Seal to door surround interface for potential leakage. • Door surround surface for damage which would prevent good sealing. <p>NOTE: This test is only applicable to airplanes with a baggage pannier in place of the ventral fuel tank OR 750 airplanes with external baggage unit.</p>	C3	1600 Flight Hrs	4-Nov-11	6,068.40	-	-	7668.40	-	N/A	N/A
520004	<p>Main entry door "not locked" warning microswitches.</p> <ul style="list-style-type: none"> • Energize the dc busbars (AMM 12-24-00). • With the main entry door open, operate the "door not closed" microswitch. • Make sure the "ENT DOOR UNLOCKED" annunciator on the Master warning System panel remains on. <p>NOTE: If the annunciator goes off, then investigate the 'door lock' microswitch for seizure in the operated position.</p> <ul style="list-style-type: none"> • Release the "door not closed" microswitch. • De-energize the dc busbars. 	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
520005	Escape hatch release mechanism. Check the operating load using the external push button (AMM 52-20-00).	B6	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
520006	Escape hatch seals. Inspect for total seal contact with airframe structure to prevent ingress of rainwater when the cabin is unpressurized.	B6	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
520007	<p>Entry door:</p> <ul style="list-style-type: none"> • Inspect seal for deterioration and damage. • Check internal handle operating loads (AMM 52-11-00). 	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
520008	<p>12 years then every 8 years.</p> <p>Main entry door.</p> <p>Access: Refer to NTM Technique.</p> <ul style="list-style-type: none"> • Comply with instructions in NTM 52-80-102. 	Structural	144 Months	30-Nov-09	5,345.40	-	28-Nov-17	-	-	-377.0	Days
520009	<p>Between 15 to 18 years, then every 8 years.</p> <p>Escape hatch.</p> <p>Access: Trim removed.</p> <ul style="list-style-type: none"> • Detail inspect the structure internally and externally for cracks, signs of corrosion and the deterioration of the protective treatment. 	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
520010	<p>Main Entry Door - Folding Handrail . Inspect:</p> <ul style="list-style-type: none"> • 800XP and 800XP Pro Line 21 airplanes (MMA 52-15-00, 601). 	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
520011	<p>External Baggage Unit (EBU) loading Door (750 Airplanes)</p> <ul style="list-style-type: none"> • Do a detailed visual inspection of the baggage door latch and pin assemblies. 	B7	800 Flight Hrs	1-Nov-11	6,068.40	-	-	0.00	-	N/A	N/A
53	Fuselage										

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
530001	8 years, then every 4 years. Skin exterior under the dorsal intake, forward of Frame 19. Access: Dorsal intake removed. • Detail inspect for cracks, signs of corrosion and deterior of the protective treatment. • Detail inspect for chafing rivets	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
530002	8 years, then every 4 years. Fuselage skin under wing-to-fuselage fairings. Access: Panels F124 and F224 removed. • Inspect visible areas of skin for cracks and signs of corrosion using flashlight, mirror and borescope probes.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
530002	8 years, then every 4 years. Fuselage skin under wing-to-fuselage fairings. Access: Panels F124 and F224 removed. • Inspect visible areas of skin for cracks and signs of corrosion using flashlight, mirror and borescope probes.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
530003	Between 15 to 18 years, then every 8 years. Fuselage skin covered by the ventral fuel tank OR baggage pannier/EBU (if fitted). Access: Ventral fuel tank OR baggage pannier/EBU (if fitted) removed. • Detail inspect for signs of corrosion and deterioration of protective treatments. • Detail inspect attachment points and links.	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
530004	Between 15 to 18 years, then every 8 years. Ventral fuel tank and attachments, . Access: Ventral fuel tank (AMM 28-10-15) and tank access panels removed. • Detail inspect the tank internal structure, skin, access panels, bolts and fasteners, fuel system components and pipelines. • Pay particular attention to the security of the nylon vent pipes and the integrity of the sealant fillets at the pipe joints. • Inspect the rubber drain non-return valve in the vent system balance pipe for deterioration and splitting. • Detail inspect the fuel seal block assembly (P/N: 25-8PT341-1A) at the ventral tank diaphragm for missing sealer, voids and corrosion. NOTE: If missing sealer, voids or corrosion are found, remove the fuel seal block assembly for further detailed inspection. • Detail inspect attachment points and links.	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
530007	12 years and then every 8 years. Wing to fuselage attachment links, brackets and bolts. Access: Panels F124, F224 and W501 removed. • Detail inspect links, brackets and bolts for cracks, signs of corrosion, fretting and wear (AMM 53-40-01).	Structural	144 Months	30-Nov-09	5,365.40	-	28-Nov-17	-	-	-377.0	Days
530007	12 years and then every 8 years. Wing to fuselage attachment links, brackets and bolts. Access: Panels F124, F224 and W501 removed. • Detail inspect links, brackets and bolts for cracks, signs of corrosion, fretting and wear (AMM 53-40-01).	Structural	144 Months	30-Nov-09	5,365.40	-	28-Nov-17	-	-	-377.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
530009	27 000 flights and then every 4500 flights. Fuselage lap joints. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-104.	Structural	27000 Ldgs	27-Jul-97	-	-	-	-	27000	21781.0	Cyl
530011	Every 4 years. Front face of rear pressure bulkhead (Frame 19). • Inspect left-hand side of bulkhead with the soundproofing blankets removed. NOTE: When installing local furnishings, make sure there is a minimum clearance of 1.0 inch between the motor of the fa operated venturi and the local furnishings. • Look for signs of corrosion at the boundary angle to Frame 19 joints. NOTE: If you find any defects, extend the inspection to include the right hand side of the bulkhead.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
530012	Every 8 years. Front face of rear pressure bulkhead (Frame 19). • Inspect right hand side of bulkhead with soundproofing blanket removed. • Look for signs of corrosion at the boundary angle to frame 19 joints. NOTE When installing local furnishings, make sure a minimum clearance of 1.0 inch between the motor of the fan operated venturi and local furnishings is kept.	Structural	96 Months	2-Feb-14	6,475.70	3,622	31-Jan-22	-	-	1148.0	Days
530013	Every 10,000 Flying Hours Pylon engine main mounting attachment fittings. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-110.	Structural	10000 Flight Hrs	27-Jul-97	-	-	-	10000.00	-	2168.9	Hrs
530013	Every 10,000 Flying Hours Pylon engine main mounting attachment fittings. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-110., Ref 53-10-110	Structural	10000 Flight Hrs	27-Jul-97	-	-	-	10000.00	-	2168.9	Hrs
530014	Every 20 000 flying hours. Rear engine mount fittings. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-110A.	Structural	20000 Flight Hrs	27-Jul-97	-	-	-	20000.00	-	12168.9	Hrs
530014	Every 20 000 flying hours. Rear engine mount fittings. Access: Refer to NTM Technique. • Comply with instructions in NTM 53-10-110A.	Structural	20000 Flight Hrs	27-Jul-97	-	-	-	20000.00	-	12168.9	Hrs
530015	Every 30 000 flying hours. Vertical stabilizer root attachment to fuselage. Access: Refer to NTM Technique. • Comply with instructions in NTM 55-30-104. Every	Structural	30000 Flight Hrs	27-Jul-97	-	-	-	30000.00	-	22168.9	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
530016	<p>Every 10 000 flying hours. Engine beam. Access: Engines and engine main mounting attachment fittings removed (AMM Chapter 71). NOTE: The pylon firewall below the engine front mount assembly and the inside of the engine pick-up mount bracket can be visually inspected from inside the pylon access panels. The ends of the engine beam with all the doublers and angles can be inspected from inside the rear equipment bay area.</p> <ul style="list-style-type: none"> Detail inspect the multi-layer ends of the beam forming pick-ups for damage, cracks and signs of corrosion using X 10 magnification. 	Structural	10000 Flight Hrs	27-Jul-97	-	-	-	10000.00	-	2168.9	Hrs
530016	<p>Every 10 000 flying hours. Engine beam. Access: Engines and engine main mounting attachment fittings removed (AMM Chapter 71). NOTE: The pylon firewall below the engine front mount assembly and the inside of the engine pick-up mount bracket can be visually inspected from inside the pylon access panels. The ends of the engine beam with all the doublers and angles can be inspected from inside the rear equipment bay area.</p> <ul style="list-style-type: none"> Detail inspect the multi-layer ends of the beam forming pick-ups for damage, cracks and signs of corrosion using X 10 magnification. 	Structural	10000 Flight Hrs	27-Jul-97	-	-	-	10000.00	-	2168.9	Hrs
530017	<p>Whenever the brackets are removed from the engine. NOTE: This task is only required when the engine has accomplished a Major Periodic Inspection. Refer to the relevant Honeywell Light Maintenance Manual: 750, 800XP and 800XP Pro Line 21 - TFE731-5BR-1H Series engine. 900XP - TFE731-50R-1H Series engine. Forward engine mounting brackets, Part No. 3071071-3. Access: Engine cowlings M1L, M2R, M3L and M4R removed.</p> <ul style="list-style-type: none"> Remove the mounting brackets. Inspect for cracks using a magnetic particle inspection (LMM 72-00-06). 	Structural	2100 Eng Hrs	10-Oct-11	5,979.00	-	-	8079.00	-	337.3	Eng Hrs
530017	<p>Whenever the brackets are removed from the engine. RH NOTE: This task is only required when the engine has accomplished a Major Periodic Inspection. Refer to the relevant Honeywell Light Maintenance Manual: 750, 800XP and 800XP Pro Line 21 - TFE731-5BR-1H Series engine. 900XP - TFE731-50R-1H Series engine. Forward engine mounting brackets, Part No. 3071071-3. Access: Engine cowlings M1L, M2R, M3L and M4R removed.</p> <ul style="list-style-type: none"> Remove the mounting brackets. Inspect for cracks using a magnetic particle inspection (LMM 72-00-06). 	Structural	2100 Eng Hrs	21-Dec-16	7,523.40	4,866	-	9623.40	-	1967.6	Eng Hrs
530018	<p>4800 flying hours, then every 2400 flying hours. Fuselage skin adjacent to cabin window cutouts. Detail inspect skin.</p>	Structural	2400 Flight Hrs	4-Nov-11	6,068.40	-	-	8468.40	-	637.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
530019	<p>PATS APU GTCP36-150W. Every 12 months.</p> <p>NOTE: If any of the following inspections show an abnormal indication or requires the replacement of structural parts, do Tasks 530022 and 530023.</p> <p>Detail inspect the following items accessible without removal from the rear equipment bay, for cracks or failures:</p> <ul style="list-style-type: none"> • All APU support structure external to the fire enclosure. • Remove the oil filler door and all access covers in the bottom enclosure and inspect all the APU structure internal to the fire enclosure. • Inspect the main engine crossbeam and frame 21 in the areas immediately adjacent to the APU attachment fittings. • Inspect the five pairs of isolator mounts for deterioration of the silicone rubber. <p>Visually check on the exterior of the fuselage and make sure that the APU exhaust duct is centered in the exhaust shroud duct, and that the exhaust shroud duct is centered in the skin cutout for the APU exhaust.</p>	Structural	12 Months	11-Aug-17	7,709.80	5,131	11-Aug-18	-	-	-121.0	Days
530021	<p>PATS APU GTCP36-150W. Every 3000 APU running hours.</p> <p>APU Support Structure:</p> <ul style="list-style-type: none"> • Detail inspect all mounting parts for cracks or damage. • Detail inspect all fuselage attachments point for cracks or damage. <p>NOTE: Repair as required.</p>	Structural	3000 APU Hrs	20-Aug-15	4,494.00	4,329	-	7494.00	-	2609.0	APU Hrs
530022	<p>PATS APU GTCP36-150W. Every 4500 APU running hours.</p> <p>NOTE Do this task in conjunction with the removal of the APU for the Hot Section inspection (Out of Phase Task # 490006).</p> <ul style="list-style-type: none"> • Remove and detail inspect the items listed in Table 1 (Located after Area F4 inspections). <p>NOTE Replace all parts with the evidence of any damage.</p>	Structural	4500 APU Hrs	20-Aug-15	4,494.00	4,329	-	8994.00	-	4109.0	APU Hrs
530023	<p>PATS APU GTCP36-150W. Every 4500 APU running hours.</p> <p>NOTE: Do this task in conjunction with the removal of the APU for the Hot Section inspection (Out of Phase Task # 490006).</p> <ul style="list-style-type: none"> • Detail inspect the fuselage structure for cracks, particularly the support bracket attachment holes in the upper cap of the main engine cross beam and at the top and left side of the fuselage at frame 21. <p>NOTE: The attachment brackets secured to the airframe will have to be removed so that a detailed inspection of the holes drilled in the airframe for the APU fitting can be done.</p>	Structural	4500 APU Hrs	20-Aug-15	4,494.00	4,329	-	8994.00	-	4109.0	APU Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
530024	PATS APU GTCP36-150W. Every 22,500 APU running hours. APU support hardware listed in Table 2 (Located after Area F4 inspections) • Do a fluorescent penetrant inspection (FPI) (Ref. NTM 125, General Techniques, Part C Liquid Penetrant) OR • Replace. NOTE: FPI using MIL-1-25135C, Group 6A penetrant. Reject all linear indications. Do this task in conjunction with the removal of the APU for the Hot Section inspection (Out of Phase Task # 490006).	Structural	22500 Flight Hrs	1-Nov-97	-	-	-	22500.00	-	14668.9	Hrs
530025	PATS APU GTCP36-150W. Every 10,000 flights. Exhaust and inlet doubler frames and stringers. • Detail inspect internally and externally for minute cracking conditions or other defects particularly near the exhaust. NOTE: Repair as required.	Structural	10000 Ldgs	2-Oct-97	-	-	-	-	10000	4781.0	Cyl
530027	Fuselage Skin: • Do a visual inspection.	G	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
53-AD	Prevent Reduced Structural Integrity Of The Fuselage & Subsequent Decompression Of The Aircraft	AD	4000 Ldgs	20-Dec-14	6,756.42	3,974	-	-	7974	2755.0	Cyl
55 Stabilisers											
550001	Every 4 years. Elevators: • Remove elevator (AMM 27-30-22). • Detail inspect. • Inspect the bearings for satisfactory condition and make sure there is adequate lubrication (AMM 125/H-20, 20-09-22, 201). • Detail inspect the hinge brackets for security and cracks with a X 10 magnifier. Look for signs of corrosion. • Install elevator (AMM 27-30-22).	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550001	Every 4 years. Elevators: • Remove elevator (AMM 27-30-22). • Detail inspect. • Inspect the bearings for satisfactory condition and make sure there is adequate lubrication (AMM 125/H-20, 20-09-22, 201). • Detail inspect the hinge brackets for security and cracks with a X 10 magnifier. Look for signs of corrosion. • Install elevator (AMM 27-30-22).	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550002	Every 4 years. Elevator trim tabs, connecting rods: • Lubricate or replace the rod end bearings, as required (AMM 125/H-20, 20-09-22, 201). • Apply a coat of silicone lubricant to both exterior surfaces of the bearing at each connecting rod end (AMM 27-30-00). NOTE: The silicone lubricant is applied to give the rod end bearing protection against corrosion.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
550002	Every 4 years. Elevator trim tabs, connecting rods: • Lubricate or replace the rod end bearings, as required (AMM 125/H-20, 20-09-22, 201). • Apply a coat of silicone lubricant to both exterior surfaces of the bearing at each connecting rod end (AMM 27-30-00). NOTE: The silicone lubricant is applied to give the rod end bearing protection against corrosion.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550003	8 years, then every 4 years. Elevators. Access: Refer to relevant NTM Technique. • Comply with instructions in NTM 55-20-101 and NTM 55-20-102. • If corrosion is found extend the X-Ray coverage until a complete rib bay is found to be free from corrosion.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550003	8 years, then every 4 years. Elevators. Access: Refer to relevant NTM Technique. • Comply with instructions in NTM 55-20-101 and NTM 55-20-102. • If corrosion is found extend the X-Ray coverage until a complete rib bay is found to be free from corrosion.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550004	Every 4 years. Rudder: • Remove (AMM 27-20-23). • Detail inspect. • Inspect the bearings for satisfactory condition and make su there is adequate lubrication (AMM 125/H-20, 20-09-22, 201). • Install (AMM 27-20-23).	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550005	Every 4 years. Connecting rods, rudder trim tab and anti-servo tab: • Lubricate or replace the rod end bearings, as required (AMM 125/H-20, 20-09-22, 201).	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550006	8 years, then every 4 years. Rudder. Access: Refer to NTM Technique. • Comply with instructions in NTM 55-40-101.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
550007	Every 8 years. Rudder hinge brackets and attachments. Access: Rudder and rudder hinge bracket cover plates removed. • Detail inspect the rudder hinge brackets, using a X 10 magnifier.	Structural	96 Months	2-Feb-14	6,475.70	3,622	31-Jan-22	-	-	1148.0	Days
550008	Every 1200 flying hours or 2 years, whichever is the sooner. Horizontal stabilizer to vertical stabilizer rear attachment fitting. • Inspect for signs of corrosion, paying special attention to the condition of the outboard flanges and webs. NOTE: For corrosion limits refer to Hawker Beechcraft Corporation.	F8	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
550009	<p>Every 8 years. Horizontal stabilizer. Access: Leading edges removed. Elevators removed. Shroud panels T21R, T22L, T23R, T24L, T25R, T26L and T45 removed. Fairing panels T4L, T4R, T5, T6L, T7R, T8L, T9R and T13 removed.</p> <ul style="list-style-type: none"> Detail inspect the front face of the front spar for cracks and signs of corrosion. Detail inspect the rear face of the rear spar for cracks and signs of corrosion. Detail inspect the vertical stabilizer to horizontal stabilizer attachment fittings. Detail inspect the elevator hinge brackets. 	Structural	96 Months	2-Feb-14	6,475.70	3,622	31-Jan-22	-	-	1148.0	Days
550009	<p>Every 8 years. Horizontal stabilizer. Access: Leading edges removed. Elevators removed. Shroud panels T21R, T22L, T23R, T24L, T25R, T26L and T45 removed. Fairing panels T4L, T4R, T5, T6L, T7R, T8L, T9R and T13 removed.</p> <ul style="list-style-type: none"> Detail inspect the front face of the front spar for cracks and signs of corrosion. Detail inspect the rear face of the rear spar for cracks and signs of corrosion. Detail inspect the vertical stabilizer to horizontal stabilizer attachment fittings. Detail inspect the elevator hinge brackets. 	Structural	96 Months	2-Feb-14	6,475.70	3,622	31-Jan-22	-	-	1148.0	Days
550010	<p>Between 15 to 18 years, then every 8 years. Vertical stabilizer. Access: Leading edge panels, upper vertical stabilizer and rudder removed.</p> <ul style="list-style-type: none"> Detail inspect the structure rendered visible. Detail inspect the rudder hinge brackets for security, cracks and signs of corrosion. Comply with instructions in: <ul style="list-style-type: none"> NTM 55-30-101 NTM 55-30-102 NTM 55-30-103. With the leading edge removed detail inspect the internal structure of the vertical stabilizer for cracks and signs of corrosion using a borescope probe through the access holes in the front spar and through the root rib 	Structural	216 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
550011	<p>16 years, then every 8 years (see NOTE). Horizontal stabilizer to vertical stabilizer attachments. Access: Horizontal stabilizer removed. NOTE: Airplanes which have had the horizontal stabilizer removed at or subsequent to the 12 year inspection can be considered to have satisfied the initial inspection requirement.</p> <ul style="list-style-type: none"> DETAIL INSPECT THE ATTACHMENT LUGS FOR CRACKS, SIGNS OF CORROSION AND ELONGATION OF THE ATTACHMENT BOLT HOLES. INSPECT THE BOLTS FOR SIGNS OF WEAR OR CORROSION. REPLACE ANY WORN BOLTS. IF HOLES ARE ELONGATED REPLACE BOLTS, SEE SRM 55-10-41. 	Structural	192 Months	24-Jul-14	6,626.40	3,798	22-Jul-22	-	-	1320.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
550011	<p>16 years, then every 8 years (see NOTE). Horizontal stabilizer to vertical stabilizer attachments. Access: Horizontal stabilizer removed. NOTE: Airplanes which have had the horizontal stabilizer removed at or subsequent to the 12 year inspection can be considered to have satisfied the initial inspection requirement.</p> <ul style="list-style-type: none"> • DETAIL INSPECT THE ATTACHMENT LUGS FOR CRACKS, SIGNS OF CORROSION AND ELONGATION OF THE ATTACHMENT BOLT HOLES. • INSPECT THE BOLTS FOR SIGNS OF WEAR OR CORROSION. • REPLACE ANY WORN BOLTS. IF HOLES ARE ELONGATED REPLACE BOLTS, SEE SRM 55-10-41. 	Structural	192 Months	24-Jul-14	6,626.40	3,798	22-Jul-22	-	-	1320.0	Days
550012	<p>24 years and then every 8 years. Horizontal stabilizer. Access: Refer to relevant NTM Techniques. NOTE: For convenience and improved access it is recommended that this inspection is done when the horizontal stabilizer is removed for the inspection of the attachment lugs (Task No. 550011).</p> <ul style="list-style-type: none"> • Comply with instructions given in one of the methods be • Fokker Bond Test, NTM 55-10-104A, • Schlumberger Sonatest NTM 55-10-104B • Staveley Sonic Bondmaster NTM 55-10-104C. Ref:55-10-104 	Structural	288 Months	6-Feb-06	3,855.70	-	31-Jan-30	-	-	4070.0	Days
550012	<p>24 years and then every 8 years. Horizontal stabilizer. Access: Refer to relevant NTM Techniques. NOTE: For convenience and improved access it is recommended that this inspection is done when the horizontal stabilizer is removed for the inspection of the attachment lugs (Task No. 550011).</p> <ul style="list-style-type: none"> • Comply with instructions given in one of the methods be • Fokker Bond Test, NTM 55-10-104A, • Schlumberger Sonatest NTM 55-10-104B • Staveley Sonic Bondmaster NTM 55-10-104C. Ref:55-10-104 	Structural	288 Months	27-Jul-97	-	-	21-Jul-21	-	-	954.0	Days
550013	<p>Between 15 to 18 years, then every 8 years. Horizontal stabilizer Assembly. Access: Refer to relevant NTM Techniques.</p> <ul style="list-style-type: none"> • Comply with instructions in: • NTM 55-10-101 • NTM 55-10-102 • NTM 55-10-103 • NTM 55-10-105. 	Structural	180 Months	23-Jul-15	7,067.50	4,326	18-Jul-33	-	-	5334.0	Days
550014	<p>15 000 flights and then at every X-Ray inspection of the horizontal stabilizer internal structure (Task No.550013). Horizontal stabilizer top skin "T" fitting and joint strap at the center line of the stabilizer. Access: Upper vertical stabilizer removed.</p> <ul style="list-style-type: none"> • Detail inspect for cracks and signs of corrosion. <p>Ref 53-10-110</p>	Structural	15000 Ldgs	23-Jul-15	7,067.50	4,326	-	-	19326	14107.0	Cyl

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
550015	<p>Every 2 years. Elevators:</p> <ul style="list-style-type: none"> Remove mass balance weight sideplates. Remove and inspect mass balance weights, spigots and installation, look for signs of corrosion. <p>NOTE: Remove all corrosion found from the spigots to a maximum depth of 0.03 in. and paint with epoxy primer (AMM 125/H-20, 20-95-401, Item No. 465).</p> <p>NOTE: If excessive corrosion is found contact: Repair D.O.Hawker Beechcraft Corporation, 9709 East Central, P.O. Box 85 Wichita, KS 67201-0085. U.S.A.</p> <ul style="list-style-type: none"> Apply corrosion preventive compound LPS 3 (AMM125/H-20, 20-95-11, Item No. 089) or ACF 50 (AMM 125/H-20, 20-95-11, Item No. 090). Assemble the spigots, mass balance weights and sideplates using sealant (AMM125/H-20, 20-95-301, Item No. 314) to minimize water ingress and corrosion. <p>CAUTION: Make sure that the mass balance weights are installed on the same side that they were removed from.</p>	Structural	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
550015	<p>Every 2 years. Elevators:</p> <ul style="list-style-type: none"> Remove mass balance weight sideplates. Remove and inspect mass balance weights, spigots and installation, look for signs of corrosion. <p>NOTE: Remove all corrosion found from the spigots to a maximum depth of 0.03 in. and paint with epoxy primer (AMM 125/H-20, 20-95-401, Item No. 465).</p> <p>NOTE: If excessive corrosion is found contact: Repair D.O.Hawker Beechcraft Corporation, 9709 East Central, P.O. Box 85 Wichita, KS 67201-0085. U.S.A.</p> <ul style="list-style-type: none"> Apply corrosion preventive compound LPS 3 (AMM125/H-20, 20-95-11, Item No. 089) or ACF 50 (AMM 125/H-20, 20-95-11, Item No. 090). Assemble the spigots, mass balance weights and sideplates using sealant (AMM125/H-20, 20-95-301, Item No. 314) to minimize water ingress and corrosion. <p>CAUTION: Make sure that the mass balance weights are installed on the same side that they were removed from.</p>	Structural	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
56	Windows	0									
560004	<p>Windows dry-air system filter (if fitted).</p> <ul style="list-style-type: none"> Inspect the filter element (AMM 56-00-00). 	C6	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
560005	<p>Passenger cabin windows:</p> <ul style="list-style-type: none"> Satisfactory condition of seals and no signs of moisture ingress. If you find moisture, renew Silica Gel crystals and verify seal integrity (AMM 56-00-00). <p>NOTE: Pre Mod. 25G360A ONLY.</p>	B3	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
560006	<p>Windshield panel 'C' Silica Gel container:</p> <ul style="list-style-type: none"> Inspect Silica Gel crystals for discoloration. Renew as necessary. 	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
57	Wings	0									

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570001	Flap hinge fittings. Inspect. NOTE: Inspect the bolt head on the bottom skin of the flap (in the area of the inboard hinge) adjacent to flap screw jack carriagegrease fitting for damage caused by chafing. If damage is found, replace the bolt. NOTE: For details of the repair kit refer to Hawker Beechcraft Corporation Technical Support (1-800-429-5372 OR 316-676-3140).	B9	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
570002	Flap hinge fittings. Inspect. NOTE: Inspect the bolt head on the bottom skin of the flap (in the area of the inboard hinge) adjacent to flap screw jack carriagegrease fitting for damage caused by chafing. If damage is found, replace the bolt. For details of the repair kit refer to Hawker Beechcraft Corporation Technical Support (1-800-429-5372 OR 316-676-3140).	B10	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
570003	The internal structure of the wing in the area of Ribs 1 to 4. • Detail inspect. • Look for signs microbiological contamination and corrosion. NOTE: If corrosion or microbiological contamination is found: • PROGRESSIVELY EXTEND THE DETAIL INSPECTION, BY VISUAL MEANS OR X-RAY TECHNIQUE, FORWARD OF THE CENTRE SPAR AND OUTBOARD OF RIB 4 UNTIL A CLEAN BAY RIB IS FOUND. • REMOVE THE VENTRAL FUEL TANKS(IF FITTED) AND DETAIL INSPECT INTERNALLY. • DO THE SHOCK PREVENTATIVE PROCEDURE (AMM 28-10-00, PAGE BLOCK 701).	F9	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	Days
570004	Applicable to Mod. 259071A airplanes only. Flap skin protection panels (if fitted). Remove panels and inspect the flap skin for corrosion and deterioration of the protective treatment. NOTE: Mod. 259071A introduces protective skins on the flap inboard underside for rough runway protection.	F9	24 Months	13-Jun-12	6,196.00	3,528	13-Jun-14	-	-	N/A	N/A
570005	The internal structure of the wing in the area of Ribs 1 to 4. • Detail inspect. • Look for signs microbiological contamination and corrosion. NOTE: If corrosion or microbiological contamination is found: • Progressively extend the detail inspection, by visual means or X-Ray technique, forward of the centre spar and outboard of Rib 4 until a clean bay rib is found. • Remove the ventral fuel tanks (if fitted) and detail inspect internally. • Do the shock preventative procedure (AMM 28-10-00, page block 701)	F10	24 Months	19-Aug-16	7,523.40	4,866	19-Aug-18	-	-	-113.0	0-Jan-00
570006	Applicable to Mod. 259071A airplanes only. Flap skin protection panels (if fitted). Remove panels and inspect the flap skin for corrosion and deterioration of the protective treatment. NOTE: Mod. 259071A introduces protective skins on the flap inboard underside for rough runway protection.	F10	24 Months	13-Jun-12	6,196.00	3,528	13-Jun-14	-	-	N/A	N/A

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570007	<p>8 years, then every 4 years. Wing bottom skins and stringers between the front and rear spars, Rib 0 to Rib 1 and between the center spar and rear spar from Rib 1 to Rib 4. Access: Refer to NTM Technique.</p> <ul style="list-style-type: none"> Comply with instructions in NTM 57-30-101. Detail inspect the bottom stringers in the vicinity of the wing skin restraint ties with the aid of a flashlight and mirrors. <p>NOTE: If corrosion is found while doing NTM 57-30-101, extend the X-Ray coverage (NTM 57-30-102) until a complete rib bay is found to be free from corrosion.</p> <p>NOTE: If the quality of the radiographs is impaired by the presence of pockets of fuel it would indicate that the internal fuel drain paths are blocked. Operators are advised to remedy this condition at the first convenient opportunity.</p>	Structural	96 Months	24-Jul-14	6,626.40	3,798	23-Jul-18	-	-	-140.0	Days
570007	<p>8 years, then every 4 years. Wing bottom skins and stringers between the front and rear spars, Rib 0 to Rib 1 and between the center spar and rear spar from Rib 1 to Rib 4. Access: Refer to NTM Technique.</p> <ul style="list-style-type: none"> Comply with instructions in NTM 57-30-101. Detail inspect the bottom stringers in the vicinity of the wing skin restraint ties with the aid of a flashlight and mirrors. <p>NOTE: If corrosion is found while doing NTM 57-30-101, extend the X-Ray coverage (NTM 57-30-102) until a complete rib bay is found to be free from corrosion.</p> <p>NOTE: If the quality of the radiographs is impaired by the presence of pockets of fuel it would indicate that the internal fuel drain paths are blocked. Operators are advised to remedy this condition at the first convenient opportunity.</p>	Structural	96 Months	24-Jul-14	6,626.40	3,798	23-Jul-18	-	-	-140.0	Days
570008	<p>8 years, then every 4 years. Main Landing Gear and sidestay attachment fittings. Access: Panels W324, W424 removed.</p> <ul style="list-style-type: none"> Detail inspect the landing gear pick-up fittings, specially the sidestay pick-up, for cracks, signs of corrosion and signs of movement. Make sure the drain holes are unobstructed. Check the tightness of all securing bolts. 	Structural	96 Months	24-Jul-14	6,626.40	3,798	23-Jul-18	-	-	-140.0	Days
570008	<p>8 years, then every 4 years. Main Landing Gear and sidestay attachment fittings. Access: Panels W324, W424 removed.</p> <ul style="list-style-type: none"> Detail inspect the landing gear pick-up fittings, specially the sidestay pick-up, for cracks, signs of corrosion and signs of movement. Make sure the drain holes are unobstructed. Check the tightness of all securing bolts. 	Structural	96 Months	24-Jul-14	6,626.40	3,798	23-Jul-18	-	-	-140.0	Days
570009	<p>4800 flying hours, then every 1200 flying hours. Wing bottom skin (LH and RH) adjacent to flap outboard hinge at Rib 8. Access: Hinge fairings, panels W314 and W414, removed.</p> <ul style="list-style-type: none"> Detail inspect for cracks and signs of corrosion. 	Structural	4800 Flight Hrs	20-Aug-15	7,068.50	4,329	-	8268.50	-	437.4	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570009	4800 flying hours, then every 1200 flying hours. Wing bottom skin (LH and RH) adjacent to flap outboard hinge at Rib 8. Access: Hinge fairings, panels W314 and W414, removed. • Detail inspect for cracks and signs of corrosion.	Structural	4800 Flight Hrs	20-Aug-15	7,068.50	4,329	-	8268.50	-	437.4	Hrs
570010	Every 4800 flying hours or 8 years, whichever is the sooner. Wing front spar, forward face. Access: Wing leading edges removed. • Detail inspect for cracks, signs of corrosion and distortion, specially between Ribs 1 and 3. • Inspect leading edges for cracks, signs of corrosion, damage and integrity of the bolt holes. NOTE : When replacing the leading edges the profiles must be maintained (see SRM 57-41-00).	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	31-Jan-22	11275.70	-	1148.0	Days
570010	Every 4800 flying hours or 8 years, whichever is the sooner. Wing front spar, forward face. Access: Wing leading edges removed. • Detail inspect for cracks, signs of corrosion and distortion, specially between Ribs 1 and 3. • Inspect leading edges for cracks, signs of corrosion, damage and integrity of the bolt holes. NOTE : When replacing the leading edges the profiles must be maintained (see SRM 57-41-00).	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	31-Jan-22	11275.70	-	1148.0	Days
570011	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Wing rear spar, rear face. Access: Ailerons, aileron shrouds, flaps and airbrakes removed. • Inspect for cracks, signs of corrosion, distortion and damage. • Inspect control surface hinge brackets on the rear spar for wear, cracks, signs of corrosion and security.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570011	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Wing rear spar, rear face. Access: Ailerons, aileron shrouds, flaps and airbrakes removed. • Inspect for cracks, signs of corrosion, distortion and damage. • Inspect control surface hinge brackets on the rear spar for wear, cracks, signs of corrosion and security.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570012	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Airbrakes. Access: Airbrakes removed. • Detail inspect. • Inspect the hinges for security.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570012	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Airbrakes. Access: Airbrakes removed. • Detail inspect. • Inspect the hinges for security.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570012	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Airbrakes. Access: Airbrakes removed. • Detail inspect. • Inspect the hinges for security.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570012	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Airbrakes. Access: Airbrakes removed. • Detail inspect. • Inspect the hinges for security.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570013	Every 2400 flying hours or 4 years, whichever is the sooner. Ailerons: • Remove (AMM 27-10-31). • Detail inspect. • Inspect the bearings for satisfactory condition and make sure there is adequate lubrication (AMM 125/H-20, 20-09-22, 201). • Install (AMM 27-10-31).	Structural	2400 Flight Hrs or 48 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	8875.70	-	-312.0	Days
570013	Every 2400 flying hours or 4 years, whichever is the sooner. Ailerons: • Remove (AMM 27-10-31). • Detail inspect. • Inspect the bearings for satisfactory condition and make sure there is adequate lubrication (AMM 125/H-20, 20-09-22, 201). • Install (AMM 27-10-31).	Structural	2400 Flight Hrs or 48 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	8875.70	-	-312.0	Days
570014	2400 flying hours or 4 years, whichever is the sooner Connecting rods, aileron trim and servo tabs: • Lubricate or replace the front and rear rod end bearings, as required (AMM 125/H-20, 20-09-22, 201). • Apply a coat of silicone lubricant to both exterior surfaces of the bearing at each trim tab connecting rod end (AMM 27-10-00). NOTE: The silicone lubricant is applied to give the rod end bearing protection against corrosion	Structural	2400 Flight Hrs or 48 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	8875.70	-	-312.0	Days
570014	2400 flying hours or 4 years, whichever is the sooner Connecting rods, aileron trim and servo tabs: • Lubricate or replace the front and rear rod end bearings, as required (AMM 125/H-20, 20-09-22, 201). • Apply a coat of silicone lubricant to both exterior surfaces of the bearing at each trim tab connecting rod end (AMM 27-10-00). NOTE: The silicone lubricant is applied to give the rod end bearing protection against corrosion	Structural	2400 Flight Hrs or 48 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	8875.70	-	-312.0	Days
570015	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Ailerons. Access: Refer to NTM Technique. • Detail inspect the hinge brackets with a X 10 magnifier. • Comply with instructions in NTM 57-50-107.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	8875.70	-	-312.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570015	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Ailerons. Access: Refer to NTM Technique. • Detail inspect the hinge brackets with a X 10 magnifier. • Comply with instructions in NTM 57-50-107.	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	8875.70	-	-312.0	Days
570016	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Flaps and flap vanes. Access: Refer to relevant NTM Technique. • Inspect hinge brackets for security, freedom from cracks and signs of corrosion. • Inspect bearings for satisfactory condition and adequate lubrication. • Comply with instructions in NTM 57-50-105 and NTM 57-50-106. NOTE: If corrosion is found during the X-Ray inspection, extend the X-Ray inspection coverage until a complete rib bay is found free of corrosion	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570016	4800 flying hours or 8 years, whichever is the sooner, then every 2400 flying hours or 4 years. Flaps and flap vanes. Access: Refer to relevant NTM Technique. • Inspect hinge brackets for security, freedom from cracks and signs of corrosion. • Inspect bearings for satisfactory condition and adequate lubrication. • Comply with instructions in NTM 57-50-105 and NTM 57-50-106. NOTE: If corrosion is found during the X-Ray inspection, extend the X-Ray inspection coverage until a complete rib bay is found free of corrosion	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	24-Jul-14	6,626.40	3,798	23-Jul-18	9026.40	-	-140.0	Days
570017	24 000 flying hours and then every 4800 flying hours. Wing bottom skin at attachment to front spar, between Ribs 1 and 7. Access: As visible. • Detail inspect for cracks and signs of corrosion.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570017	24 000 flying hours and then every 4800 flying hours. Wing bottom skin at attachment to front spar, between Ribs 1 and 7. Access: As visible. • Detail inspect for cracks and signs of corrosion.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570018	24 000 flying hours and then every 9600 flying hours. Wing bottom skin centerline joint strap. Access: Keel skid removed. • Detail inspect for cracks and signs of corrosion.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570018	24 000 flying hours and then every 9600 flying hours. Wing bottom skin centerline joint strap. Access: Keel skid removed. • Detail inspect for cracks and signs of corrosion.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570019	24 000 flying hours and then every 9600 flying hours. Upper surface of the wing bottom skin at the attachment to the centerline joint adjacent to Rib 0. Access: Refer to NTM Technique. • Comply with instructions in NTM 57-30-104. OR 24 000 flying hours and then every 7200 flying hours. • With the sealant removed to expose the skin surface, detail inspect for cracks at the bolt line.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570019	24 000 flying hours and then every 9600 flying hours. Upper surface of the wing bottom skin at the attachment to the centerline joint adjacent to Rib 0. Access: Refer to NTM Technique. • Comply with instructions in NTM 57-30-104. OR 24 000 flying hours and then every 7200 flying hours. • With the sealant removed to expose the skin surface, detail inspect for cracks at the bolt line.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570020	24 000 flying hours and then every 12 000 flying hours. Wing bottom skin centerline butt strap and center section beams (24 positions). Access: Refer to NTM Technique. • Comply with instructions in NTM 57-00-102.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570020	24 000 flying hours and then every 12 000 flying hours. Wing bottom skin centerline butt strap and center section beams (24 positions). Access: Refer to NTM Technique. • Comply with instructions in NTM 57-00-102.	Structural	24000 Flight Hrs	30-Nov-09	5,345.40	-	-	29345.40	-	21514.3	Hrs
570022	Every 4800 flying hours. Rear spar, rear face between Rib 1 LH and Rib 1 RH. Access: Main landing gear doors and panels W325, W425, W507 and W508 removed. • Detail inspect the rear face of the rear spar.	Structural	4800 Flight Hrs	6-Feb-06	3,854.60	-	-	8654.60	-	823.5	Hrs
570023	Every 4 years. NOTE: Applicable to airplanes without HBC Winglets. Aileron horn balance weight. Access: Aileron horn balance weight cover removed. • Detail inspect the horn balance weight for signs of corrosion. • If signs of corrosion are found, then do an X-Ray inspection of the area • Comply with instructions in NTM 57-50-107.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
570024	12 000 flying hours, then every 7200 flying hours. Wing bottom skin at attachment to Rib 1 LH and RH. Access: As visible. • Detail inspect the skin where it is bolted to Rib 1 LH and RH.	Structural	12000 Flight Hrs	27-Jul-97	-	-	-	12000.00	-	4168.9	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570024	12 000 flying hours, then every 7200 flying hours. Wing bottom skin at attachment to Rib 1 LH and RH. Access: As visible. • Detail inspect the skin where it is bolted to Rib 1 LH and RH.	Structural	12000 Flight Hrs	27-Jul-97	-	-	-	12000.00	-	4168.9	Hrs
570025	Every 4800 flying hours. Wing bottom skin at attachment to the rear spar, between Rib 1 LH and Rib 1 RH. Access: Main landing gear doors open and keel skid fairing in the area of the main landing gear doors removed. • Detail inspect the skin where it is bolted to the rear spar, between Rib 1 LH and Rib 1 RH.	Structural	4800 Flight Hrs	6-Feb-06	3,854.60	-	-	8654.60	-	823.5	Hrs
570025	Every 4800 flying hours. Wing bottom skin at attachment to the rear spar, between Rib 1 LH and Rib 1 RH. Access: Main landing gear doors open and keel skid fairing in the area of the main landing gear doors removed. • Detail inspect the skin where it is bolted to the rear spar, between Rib 1 LH and Rib 1 RH.	Structural	4800 Flight Hrs	6-Feb-06	3,854.60	-	-	8654.60	-	823.5	Hrs
570026	Every 4 years. Wing leading edges. Access: TKS panels removed (also any material that might interfere with visual inspection). Inspect, specially for cracks and signs of corrosion.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days
570026	Every 4 years. Wing leading edges. Access: TKS panels removed (also any material that might interfere with visual inspection). Inspect, specially for cracks and signs of corrosion.	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	-	-	-312.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
570027	<p>Every 4800 flying hours or 8 years, whichever is the sooner. Wing internal structure forward of the center spar including the front and center spars from Ribs 1 to 4, and between and including the front and rear spars from Rib 4 to the wing tip. Access: Wing leading edges W102, W120, W130, W202, W220,W230 W220,W230 removed. Front spar access panels W301, W302, W303, W304, W305, W401, W402, W403, W404, W405 removed. Rear spar access panels W114, W115, W116, W117, W214, W215, W216, W217 removed. Top skin panels W106, W108, W112, W119, W206, W208, W212, W219 removed. Bottom skin panels W308, W309, W310, W312, W329, W330, W408, W409, W410, W412, W429, W430 removed.</p> <ul style="list-style-type: none"> • Detail inspect the internal structure, as visible. • Inspect for signs of corrosion and microbiological contamination. • Detail inspect the bottom stringers in the vicinity of the wing skin restraint ties with the aid of a flashlight and mirrors. <p>NOTE: If corrosion or contamination is found, clean and repair in accordance with the Structural Repair Manual. Do the shock preventative treatment in accordance with AMM 28-10-00. Fuel system non-return valves in inner and outer wings:</p> <ul style="list-style-type: none"> • Remove and inspect (AMM 28-20-32). • Pay particular attention to the condition of the valve and valve seat. 	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	31-Jan-22	11275.70	-	1148.0	Days
570027	<p>Every 4800 flying hours or 8 years, whichever is the sooner. Wing internal structure forward of the center spar including the front and center spars from Ribs 1 to 4, and between and including the front and rear spars from Rib 4 to the wing tip. Access: Wing leading edges W102, W120, W130, W202, W220,W230 W220,W230 removed. Front spar access panels W301, W302, W303, W304, W305, W401, W402, W403, W404, W405 removed. Rear spar access panels W114, W115, W116, W117, W214, W215, W216, W217 removed. Top skin panels W106, W108, W112, W119, W206, W208, W212, W219 removed. Bottom skin panels W308, W309, W310, W312, W329, W330, W408, W409, W410, W412, W429, W430 removed.</p> <ul style="list-style-type: none"> • Detail inspect the internal structure, as visible. • Inspect for signs of corrosion and microbiological contamination. • Detail inspect the bottom stringers in the vicinity of the wing skin restraint ties with the aid of a flashlight and mirrors. <p>NOTE: If corrosion or contamination is found, clean and repair in accordance with the Structural Repair Manual. Do the shock preventative treatment in accordance with AMM 28-10-00. Fuel system non-return valves in inner and outer wings:</p> <ul style="list-style-type: none"> • Remove and inspect (AMM 28-20-32). • Pay particular attention to the condition of the valve and valve seat. 	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	31-Jan-22	11275.70	-	1148.0	Days

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining		
				Date	Hours	Cycle	Date	Total Time	Total Cycle			
570028	4800 flying hours or 8 years, whichever is the sooner, then every 4 years. NOTE: Applicable to airplanes without HBC Winglets. Aileron at outboard hinge casting. • Do the "Aileron Horn" Inspection/Check (Ref. AMM 27-10-31, 601).	Structural	4800 Flight Hrs or 96 Months (Which ever is earlier)	2-Feb-14	6,475.70	3,622	1-Feb-18	11275.70	-	-312.0	Days	
570028	4800 flying hours or 8 years, whichever is the sooner, then every 4 years. NOTE: Applicable to airplanes without HBC Winglets. Aileron at outboard hinge casting. • Do the "Aileron Horn" Inspection/Check (Ref. AMM 27-10-31, 601).	Structural	48 Months	2-Feb-14	6,475.70	3,622	1-Feb-18	11275.70	-	-312.0	Days	
57-COMPT	Flap Hinge Attach Angle Bolts LH INNER	Replace	14600 Ldgs	27/07/1997	-	-	-	-	14600	9381.0	Cyl	
57-COMPT	Flap Hinge Attach Angle Bolts RH INNER	Replace	14600 Ldgs	27/07/1997	-	-	-	-	14600	9381.0	Cyl	
57-SB	N/A Winglets	SB 57-4112	600 Hrs	22-Dec-15	7,194.90	4,517	-	-	-	N/A	N/A	
72	Engine											
720002	Whenever an engine is removed for a Major Periodic Inspection. Aft cowling, inner fan duct and exhaust duct. Inspect with the aft cowling removed.	OOP	2100 Eng Hrs	10-Oct-11	5,979.00	-	-	-	8079.00	-	337.3	Eng Hrs
720002	Whenever an engine is removed for a Major Periodic Inspection. Aft cowling, inner fan duct and exhaust duct. Inspect with the aft cowling removed.	OOP	2100 Eng Hrs	19-Aug-16	7,523.40	4,866	-	-	9623.40	-	1967.6	Eng Hrs
720003	Whenever an engine is removed for a Major Periodic Inspection. Exhaust nozzle/exhaust pipe V clamp. Detail inspect, particularly for cracks in the weld lines of the trunnion nut support tubes, using a X 10 magnifier.	OOP	2100 Eng Hrs	10-Oct-11	5,979.00	-	-	-	8079.00	-	337.3	Eng Hrs
720003	Whenever an engine is removed for a Major Periodic Inspection. Exhaust nozzle/exhaust pipe V clamp. Detail inspect, particularly for cracks in the weld lines of the trunnion nut support tubes, using a X 10 magnifier.	OOP	2100 Eng Hrs	19-Aug-16	7,523.40	4,866	-	-	9623.40	-	1967.6	Eng Hrs
72-LLP	Lp 1St Stage Axial Compressor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 1St Stage Axial Compressor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 2Nd Stage Axial Compressor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 2Nd Stage Axial Compressor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 3Rd Stage Axial Compressor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 3Rd Stage Axial Compressor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Hp Shouldered Shaft LH	Replace	12000 Ldgs	4-Oct-96	-	-	-	-	12000	6781.0	Cyl	
72-LLP	Hp Shouldered Shaft RH	Replace	12000 Ldgs	28-Sep-96	-	-	-	-	12000	6781.0	Cyl	
72-LLP	Hp Compressor Radial Impeller LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Hp Compressor Radial Impeller RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Fan Rotor Disc & Hub LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Fan Rotor Disc & Hub RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Hp Turbine Wheel Rotor LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Hp Turbine Wheel Rotor RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 4Th Stage Axial Compressor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 4Th Stage Axial Compressor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 2Nd Stage Turbine Rotor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 2Nd Stage Turbine Rotor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 3Rd Stage Turbine Rotor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 3Rd Stage Turbine Rotor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 1St Stage Turbine Rotor Disc LH	Replace	10000 Ldgs	4-Oct-96	-	-	-	-	10000	4781.0	Cyl	
72-LLP	Lp 1St Stage Turbine Rotor Disc RH	Replace	10000 Ldgs	28-Sep-96	-	-	-	-	10000	4781.0	Cyl	

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
72-LLP	Hp Rotor Seal Plate LH	Replace	30000 Ldgs	4-Oct-96	-	-	-	-	30000	24781.0	Cyl
72-LLP	Hp Rotor Seal Plate RH	Replace	30000 Ldgs	28-Sep-96	-	-	-	-	30000	24781.0	Cyl
72-LLP	Fan Support AssyLH	Inspect	4200 Eng Hours	10-Dec-05	3,854.60	-	-	8054.60	-	312.9	Eng Hrs
72-LLP	Fan Support Assy RH	Inspect	4200 Eng Hours	19-Aug-16	7,348.10	4,783	-	11548.10	-	3892.3	Eng Hrs
72-AD	To Prevent Failure Of The Hpc Impeller P/N 3073393-1,3073394-1,3073433-1, 3073434-1,3073398-All,3073435-All	AD	4200 Eng Hrs	12-Jan-06	3,855.00	-	-	8055.00	-	313.3	Eng Hrs
72-AD	LH To Prevent Lpt Blade Separation	AD	2600 Eng Hr	10-Oct-11	5,979.00	-	-	8579.00	-	837.3	Eng Hrs
72-AD	RH To Prevent Lpt Blade Separation	AD	2600 Eng Hr	20-Oct-09	5,170.10	-	-	7770.10	-	114.3	Eng Hrs
72-SB	Perform Eddy Current Inspection of HP Impeller PN 3073993-1, 3073994-1, 3073433-1 & 3073434-1	SB	5000 Flight Hrs	12-Jan-06	3,855.00	1,857	-	8855.00	-	1023.9	Hrs
72-SB	Turbine Section-HPT & LPT Blade Hour Tracking	SB	2100 Eng Hrs	10-Oct-11	5,979.00	-	-	8079.00	-	337.3	Eng Hrs
72-SB	Turbine Section-HPT & LPT Blade Hour Tracking	SB	2100 Eng Hrs	19-Aug-16	7,348.10	4,783	-	9448.10	-	1792.3	Eng Hrs
72	Engine Compressor Zone InspectionLH	OOP	4200 Eng Hrs	20-Dec-05	3,854.60	1,857	-	8254.60	-	512.9	Eng Hrs
72	Engine Compressor Zone InspectionRH	OOP	4200 Eng Hrs	19-Aug-16	7,348.10	4,783	-	11748.10	-	4092.3	Eng Hrs
72-AD	LPT Roter	AD	8 Years	4-Oct-12	6,196.00	-	2-Oct-20	-	-	662.0	Days
72-AD	LPT Roter	AD	8 Years	4-Oct-12	6,196.00	-	2-Oct-20	-	-	662.0	Days
72-AD	LPT Roter	AD	4200 Eng Hrs	4-Oct-12	6,106.60	-	-	10306.60	-	2564.9	Eng Hrs
72-AD	LPT Roter	AD	4200 Eng Hrs	4-Oct-12	6,020.70	-	-	10220.70	-	2564.9	Eng Hrs
73	Engine Fuel and Control										
730001	PT2 pipelines to fuel computers. Drain any accumulations of water from the drain points.	B7	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
731120	Engine Fuel Filter Replacement LH	OOP	650 Eng Hrs	11-Aug-17	7,620.40	5,098	-	8270.40	-	528.7	Eng Hrs
731120	Engine Fuel Filter Replacement RH	OOP	650 Eng Hrs	11-Aug-17	7,534.50	5,048	-	8184.50	-	528.7	Eng Hrs
76	Engine Controls										
760001	Throttle lever controls. • Correct operation of the gust lock operated thrust lever baulk (AMM 27-72-00, page block 501).	B4	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
760002	Engine power control cables. Check cable tensions are correct (AMM 76-10-00).	C5	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
78	Exhaust										
780001	Link arm assemblies (8 off). • Remove the link arms (AMM 78-31-10). • Inspect the attachment bolts (16 off), particularly for signs of corrosion and/or seizing. • Replace any bolts that show signs of corrosion and/or seizing. • Lubricate the bolt shafts and threads before installation, using a dry film lubricant to specification MIL-PRF-907, C5-A (non-aerosol) Item 084 (AMM 20-95-11). • Install the link arms (AMM 78-31-10).	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780002	Reverser door seals. • Inspect for correct seating (AMM 78-31-20). • Inspect for cracks and wear. • Make sure the material is pliable.	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780003	Thrust reverser stow/deploy microswitches. • Test microswitches coded CWA or CXA (AMM 78-34-20, 201).	C4	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
780004	Pylon skin panels. • Remove the panels, inspect the fasteners and attachments.	C11	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
780005	Door link rods and pivot brackets. • Detail inspect (AMM 78-31-10,601).	C11	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
780006	Thrust reverser lever baulk mechanism. <ul style="list-style-type: none"> • Deploy each thrust reverser in turn. • Make sure that the baulk release lever of the reverser system which is not deployed stays down and does not show any tendency to lift when the solenoid of the deployed system is energized. • Return both thrust reverser to the stowed position. 	D4	3200 Flight Hrs	20-Aug-15	7,054.60	4,316	-	10254.60	-	2423.5	Hrs
780009	Thrust Reversers. <ul style="list-style-type: none"> • Inspect the afterbody and make sure it is correctly attached. • Inspect the doors for corrosion and wear. • Detail inspect the doors for cracks. • Inspect the fan flaps for signs of fatigue and make sure they are correctly attached and stow completely. • Detail inspect the fan flaps for cracks. 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780010	Thrust Reversers. <ul style="list-style-type: none"> • Inspect the hydraulic lines for signs of leakage. • Inspect the wire harness for any chaffing in the area of the hydraulic control valve. 	B7	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780011	Exhaust nozzle. <ul style="list-style-type: none"> • Detail inspect for cracks 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780012	Hydraulic Lines. <ul style="list-style-type: none"> • Make sure the hydraulic lines are correctly attached. • Inspect for leaks and correct routing. • Inspect the sleeving material for signs of chafing. 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780013	Wire harness. <ul style="list-style-type: none"> • Make sure the wire harness in the thrust reverser and afterbody is correctly attached. • Inspect for signs of chafing. 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780014	Fan duct. <ul style="list-style-type: none"> • Inspect for signs of corrosion and fatigue. • Detail inspect for cracks. • Inspect the contour and general condition of the fan duct fairings. 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780015	Latch box springs. <ul style="list-style-type: none"> • Inspect for signs fatigue. • Detail inspect for cracks. 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780016	Latch monitoring switches. <ul style="list-style-type: none"> • Do an operational test (AMM 78-33-50, ADJUSTMENT/TEST). 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780017	Latches and latch receptacles. <ul style="list-style-type: none"> • Inspect for wear (78-31-35, INSPECTION/CHECK) 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780018	Primary (AMM 78-32-40) and Latch (AMM 78-32-50) hydraulic actuators. <ul style="list-style-type: none"> • Inspect for leaks, corrosion or wear. • Detail inspect for cracks. 	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs

Task #	Task Details/Designation	Check	Task Intervals	Compliance			Next Due			Remaining	
				Date	Hours	Cycle	Date	Total Time	Total Cycle		
780019	Throttle Retard hydraulic actuators (AMM 78-32-30). • Inspect for leaks, corrosion or wear. • Detail inspect for cracks. • Detail inspect the actuator housing at the bearing on the Throttle Retard.	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780020	Fan Flap hydraulic actuators (AMM 78-32-60). • Inspect for leaks, corrosion or wear. • Detail inspect for cracks. • Detail inspect the actuator housing at the bearing on the Fan Flap.	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780021	Pylon Plug. • Make sure the pylon plug fairings M17L and M18R are correctly installed and make a continuous surface with the pylon	B11	800 Flight Hrs	21-Dec-16	7,523.40	4,866	-	8323.40	-	492.3	Hrs
780022	NOTE: This task is NOT APPLICABLE to series 900XP Airplanes. Door link rods. • Do a thickness check , Link rod -corrosion removal limits). 800XP airplanes (Ref. AMM 78-31-10, 601).	C11	1600 Flight Hrs	21-Dec-16	7,523.40	4,866	-	9123.40	-	1292.3	Hrs
79	Oil										
790002	ENGINE OIL CHANGE, M/JET 254	OOP	1200 Eng Hrs	22-Dec-15	7,194.90	4,517	-	8394.90	-	653.2	Eng Hrs
790002	ENGINE OIL CHANGE, M/JET 254	OOP	1200 Eng Hrs	19-Dec-16	7,523.40	4,866	-	8723.40	-	1067.6	Eng Hrs
790110	SOAP Analysis	OOP	250 Eng Hrs	11-Aug-17	7,620.40	5,098	-	7870.40	-	128.7	Eng Hrs
	END										